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JU. S. NAVAL TECHNICAL MISSION TO JAPAN CARE OF FLEET POST OFFICE

SAN FRANCISCO, CALIFORNIA

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From:

Chief, Naval Technical Mission to Japan.

To:

Chief of Naval Operations.

Subject:

Target Report - Japanese Light Armor, Article 1.

Reference: (a)"Intelligence Targets Japan" (DNI) of 4 Sept. 1945.

Subject report, dealing with Targets 0-36 and 0-37 of Fascicle 0-1 of reference (a), is submitted herewith.

The investigation of the target and the target report were accomplished by Lt. Comdr. J. J. Glancy, USNR, assisted by Lt. Comdr. M. R. Herman, USNR, and Lieut. R. R. Boggess, USNR, interpreter and translator.

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JAPANESE LIGHT ARMOR ARTICLE 1

"INTELLIGENCE TARGETS JAPAN" (DNI) OF 4 SEPT. 1945
FASCICLE O-1, TARGETS O-36, AND O-37

FEBRUARY 1946

U.S. NAVAL TECHNICAL MISSION TO JAPAN

SUMMARY

ORDNANCE TARGETS

JAPANESE LIGHT ARMOR - ARTICLE 1

This report covers light ermor used by the Japanese during World War II. Various data on steel armor are given, such as the types used, heat treatment, hardness, chemical composition, and ballistic requirements.

The Japanese used principally silicon-manganese-chromium electric-furnace steel of 0.2 to 0.4% carbon for aircraft armor. This armor was either homogeneous or carburized and was heat treated by quenching and tempering to a specific hardness. Ballistic tests conducted on the experimental and production armor are given.

Armor and structural steel in tanks, landing barges, and gun shields are discussed briefly.

The Japanese used safety glass and sponge rubber for protection of pilots and fuel cells, respectively, but there was no indication that materials such as aluminum alloy, magnesium alloy, nylon, silk, glass cloth, or leather were used as protection for personnel or equipment against the effects of minor caliber gunfire.

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REFERENCES

Location of Target:

First Naval Technical Arsenal, YOKOSUKA, and branch, KANAGAWA. Naval Aeronautical Ordnance Department, Navy Ministry, TCKYC. Naval Arsenal, JAPAN Steel Making Department, KURE. Steel Experimental Laboratory, Proving Ground, KAMEGAKUBI. Tohcku Imperial University, SENDAI. Naval Technical Research Institute, Meguro, TCKYC. Nippon Special Steel Co., Omori, TOKYC.

Japanese Fersonnel Interviewed:

- Kazuc NAKAMAE, former Technical Commander in the Steel-Making Department, Kure Naval Arsenal; heavy armor production metallurgist, 15 years experience in the manufacture of action.
- Kazuo HORIKAWA, former Lieutenant Commander in Steel-Making Experimental Laboratory, Kure Arsenal; heavy armor research metallurgist, four years experience with armor.
- Mitsuo KANERO, Civilian engineer in Experimental Ordnance Section, Kure Neval Arsenal; seven years experience in ballistic testing of armor, projectiles, and guns.
- Koi KAWAMURA, former Engineer Captain in Naval Aeronautical Ordnance Department, TOKYO: PhD in engineering; 10 years experience with aircraft ordnance.
- Todac SATO, former Technical Lieutenant Comdr. at First Naval Technical Arsenal, YOKCSUKA: research metallurgist in ferrous and non-ferrous metals used in aircraft; eight years experience.
- Seteji UNISHI, engineer at Kanagawa Branch of First Naval Technical Arsenal; six years experience in tests of aircraft machine guns and ballistic tests of aircraft armor.
- Masaichi SATO, chief production engineer at Nippon Special Steel Co.; 30 years experience in steel manufacturing, six years experience in light armor manufacture.
- T. CYAMA, engineer at Naval Technical Research Institute, Meguro, TCKYO; eight years experience in experimental testing of naval materials.
- Keisuke UDO, former Major at Army Air Force Technical Laboratory, TACHIKAWA; graduate in metallurgy, Tohoku Imperial University SENDAI; not much experience in aircraft armor.
- T. MURAKAMI, Professor of Metallurgy at Tohoku Imperial University, SENDAI; 30 years experience in metallurgical engineering.
- K. HCNDA, President of Tohoku Imperial University, SENDAI; world famous metallurgist; 40-50 years experience in metallurgy.
- I. OHJRA, former Navy Technical Captain, Ordnance Section, Navy Ministry, TCKYO.

Reports of Other Intelligence Agencies:

Technical Intelligence Co. No. 5250, Ordnance Report No. 7.

INTRODUCTION

During World War II light armor was used in tanks and aircraft by the Japanese Army and Navy. This armor was alloy steel less than two inches in thickness. Some of the details of manufacture, together with the ballistic properties, are given in this report. Most of the data describe 3mm to 16mm aircraft armor as manufactured by the Navy. Japanese tank armor and the structural type steel used for landing craft and gunshields are described briefly.

The questioning of Japanese naval aircraft ordnance personnel revealed that they did not consider aircraft armor to be effective. For example, they believed that for a given weight of armor carried in an airplane, the speed end meneuverability lost would have been of more value than the protection gained. Such views appear to have inhibited the development of Japanese aircraft armor. Nevertheless, a certain amount of aircraft armor was used. During 1943 the Japanese Navy used from 5mm to 7mm homogeneous armor as pilot-seat armor in the Zero fighter, in the Betty bember, and in torpedo bombers. During the remainder of the war, however, 12mm carburized armor was used for rear protection to the pilot in such planes as the Zero fighter, francis 11, Jill 12 and Judy.

THE REPORT

PART I - AIRCRAFT ARMCR

A. GENERAL

Although the Japanese Navy experimented with various aircraft armor steels as early as 1939, production was not undertaken until 1943. Prior to that time, the Japanese Army used aircraft armor steel, and throughout the war, used more and heavier aircraft armor than the Navy.

Information available shows that the armor steel used by the Japanese was of two types, homogeneous and carburized. It varied in thickness from 3mm to 20mm, but very little armor over 13mm in thickness was actually used. The following table lists the principal manufacturers in order of the quantity of light armor produced for Japanese Army and Navy aircraft.

Location

| | Nippon Iron Mfg. Cc. | Yawata, KYUSHU | 11.000 lbs. | |
|----|---------------------------|----------------|-------------|-------|
| ٠. | Daido Steel Mfg. Co. | NAGOYA | 11,000 lbs. | |
| | Nirpon Special Steel Co. | Omori, TCKYO | 22,000 lbs. | |
| | Nippon Steel Mfg. Co. | Muroran, | 33,000 lbs. | (only |
| | - | HOKKAIDČ | armor for | Navy) |
| | Nippon Electric Steel Co. | YCKOHAMA - | (only armor | for |
| | | | Army) | |

^{*}Monthly Average for Navy (1944-45)

Manufacturer

These manufacturers of aircraft armor used a Hercult type basic electric furnace for production armor and occasionaly a high-frequency induction furnace for experimental armor. The Hercult type furnace varied in capacities from 5 to 15 tons. Table I shows the chemical composition and specifications for production armor manufactured by the Japanese Navy.

Table I
SPECIFICATIONS AND COMFOSITION OF NAVY ARMCR

| Spec. | | Thickness (mm) | | | | | | |
|-----------|---------------|----------------|--------|---------|-------------|---------|------|----------------|
| | C, | Mn | Si | P-S | Ni | Cr | Мo | |
| Ro 601* | 0.35- | 0.8- | < 0.35 | < 0.030 | 2.5- 3.5 | 1.5-2.0 | 0.4- | 3,4,5, 6,7, |
| Ro 631** | 0.18- 0.25 | 0.8- | 0.6- | < 0.030 | | 1.0- | | 8,12 |
| Ro 632*** | 0.23- | 0.8- | 0.6- | < 0.030 | | 1.0- | | 16 |

^{*}For Homo. armor up to 1943 and for 7mm Homo. only during 1944-45.

^{**}For carburized armor only.

^{***}For carburized armor only.

During the early part of the war, armor for the Japanese Army was made of nickel-chromium-molybdenum, manganese-chromium-molybdenum, and chromium-molybdenum steels with 0.35% to 0.50% carbon.

Table II lists the specifications for analyses of armor used by the Japanese Army during 1944 and 1945.

Table II
SPECIFICATIONS AND COMPOSITION OF ARMY ARMOR

| Spec. | 1 | Thickness (mm) | | | | |
|---------|---------------|----------------|-------------|---------|---------|---------|
| - | С | Mn | si | ,P-S | Cr | |
| 2-704* | 0.35- 0.45 | 0.8- 1.8 | 0.8- 1.8 | <0.030 | 0.8- | 3 to 20 |
| 2-707** | 0.25- 0.35 | 0.9- | 0.9- 1.1 | < 0.030 | 1.0-2.0 | 7 to 20 |

^{*}For homogeneous armor.

B. MANUFACTURING PROCEDURE

1. Aircraft Armor

As stated previously, basic electric furnace steel was used for eircraft armor. Ingets produced by different manufacturers were various sizes and usually weighed from 660 to 2,200 pounds. Ingets were cropped approximately 20% from the top and 10% from the bottom. All plates were cross relied. (See Enclosure (C) for details of relling performed at the Daido Steel Mfg. Co.) All manufacturers except the Daido Steel Mfg. Co. carburized the armor in the final thickness.

2. Homogeneous Armor

The steels listed in Tables I and II were used by the Japanese Navy and Army for homogeneous armor. Plates were heat treated by normalizing at 850°C - 900°C , holding for one hour, quenching cold in oil or a press, and tempering at 100°C - 200°C for two hours.

The quenching press was used for plates under 7mm in thickness to prevent warping. The plates of the press were cooled by air or circulating water The desired hardness for 3mm to 7mm Nevel eircraft armor was from 500 Brinell to 560 Brinell while the 3mm to 16mm armor for Army aircraft ranged in hardness from 500 BHN to 630 BHN.

3. Carburized Armor

The carburizing of armor (see analysis in Tables I and II) was conducted in the final gauge except at the Daido Steel Mfg. Co. where armor was carburized in a slab 30mm in thickness, as described in Enclosure (E). The carburizing process was quite similar at all manufacturing plants. The usual commercial carburizing compounds of a mixture of berium carbonate and charcoal were used. The plates were pack carburized two at a time, face to face, in a cast steel or rolled steel box. This box was placed in a car type furnace; a carburizing temperature of 950°C was used. Carburizing time was varied according to the depth of case desired and the carburizing compound used. The aim depth of case was 20% of the nominal thickness. After carburizing, except at the Daido Steel Mfg. Co. where plates had to be rolled first to the desired thickness, the plates

^{**}For carburized armor.

were heat treated as follows: normalized at 800 - 850°C, annealed at 700°C for one hour, cooled in air, hardened at 800 - 850°C for one hour, quenched in cil tempered at 100 - 200°C for two hours. The resultant hardness of the armor was the prime factor and heat treatments were adjusted, at times, to give a face hardness greater than 550 BHN and a back hardness greater than 400 BHN.

C. INSPECTION OF ARMOR

In general, both the Army and Navy had inspectors or representatives with each of the companies. These inspectors were not always detailed to inspect light armor but, in a general manner, they guided the production of armor. Inspection was performed primarily by the company with Navy and Army light armor inspectors being summoned periodically. Apparently no hard and fast rules were set up for government inspection but the manufacturer was expected to abide by the fundamental specifications for analysis, hardness, and dimensional tolerances. The following inspection was carried out by each company to insure a satisfactory product:

- (1) Analysis check taken from the ladle
- (2) Dimensional check of each plate after completion with an allowance of plus 10% and minus 0% of the desired thickness
- (3) General-appearance check of each plate after completion
- (4) Hardness measurements on the face and back of each plate after completion
- (5) Feriodic tensile and shock tests, usually upon request of the government inspector. If a plate did not meet the hardness requirements shown in Table III, it was reheated or scrapped.

Table III
SPECIFIED MECHANICAL PROFERTIES

| Spec. | | For | Carburized Face Brinell Hardness | | |
|---|-------------------------------------|------------------------|-------------------------------------|---|----------------------------------|
| | Tensile Strength (psi) | Elongation (o/o) | Charpy (ft/lbs) | Brinell Herdness | |
| Navy Ro 601 Navy Ro 631 Navy Ro 632 Army 2-704 Army 2-707 | > 250,000 > 199,000 > 213,000 | > 6 > 8 > 5 - | >50 >36 | >500 >560 >400 >400 >500 630 >400 | >550 >550 >550 >550 |

After a group of plates, consisting of all plates from one heat of steel, had been completed, a government inspector generally was called to check hardness and dimensions, end to select a ballistic-test sample. This practice was altered in the later stages of the war and ballistic-test samples were not selected; only inspections for hardness were performed. The reduction in standard was accepted to speed up production of aircraft.

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D. BALLISTIC TESTING OF AIRCRAFT ARMOR

1. Navy Tests

The approximate conditions of test for acceptance of Naval aircraft armor are listed in Table IV. Acceptance ballistic testing was conducted by the Navy on one plate from each group. A group of plates consisted of all plates processed in the same manner and from a single heat of steel. Ballistic tests were conducted by the inspector at the plant of the manufacturer or at the branch of the First Naval Technical Arsenal of KANAGAWA. Towards the end of the war, all armor was supplied without the ballistic test.

Table IV
TEST CONDITIONS FOR JAPANESE NAVY AIRCRAFT ARMOR

| Spec. | Thickness (in.) | Projectile (AP) | Cbli- quity | Striking Vel. (ft/sec) | Remarks |
|--------|--------------------------------|----------------------------|----------------|------------------------|--|
| Ro 601 | 0.197-5mm(1) 0.236 0.276 | 7.7mm(2) 7.7mm 7.7mm | 0 0 0 | 2362 2558 2624 | Firing Range was 50 meters from gun to plate |
| Ro 631 | 0.315 0.472 0.472 | 13nm(3) 13mm 20mm(4) | 0 0 0 | 1640 2132 1476 | Impacts had to be at least three calibers apart as measured from the edge of impacts |
| Ro 632 | 0.632 0.632 | 13mm 20mm | 0 | 2378 1886 | |

- (1) Navy armor of less than 5mm thickness was not given a ballistic test. This armor was found to have no practical application in service.
- (2) An 11.4 gram projectile consisting of a 5.75 gram A.P. core with a 2.85 gram lead nose cap and a 2.80 gram copper jacket.
- (3) A 51.8 gram projectile consisting of a 31.8 gram A.P. core with a 2.9 gram lead nose cap and a 17.2 gram copper jacket.
- (4) A 116 gram A.F. projectile inert loaded to 131.3 grams for armor tests.

Complete descriptions of the projectiles referred to in notes may be found in a Japanese document entitled "Various Types of Machine-Gun Ammunition", which has been forwarded to the Washington Document Center, under NavTechJap Document No. ND50-3111, ATIS No. 4364.

In addition to the conditions of test listed in Table IV, for naval aircraft armor from 5mm to 16mm in thickness, no further velocity compensations were made for slight variations in plate thickness. At least three projectiles were fired to test a plate. Velocities were measured by a "Boulenge" chronograph or an oscillograph. As a rule, when two projectiles were listed for testing a particular thickness of armor, both were fired. If a plate passed the test with one projectile and failed with the other, the group of armor represented by the test plate would be rejected and scrapped.

A plate failed the ballistic test under the conditions set forth in Table IV when: (1) a depth of penetration was greater than 5mm, providing the striking velocity of the projectile was equal to or less than that listed; and (2) cracking or breaking of the plate was excessive, as

judged by the inspector; and (3) unless one impact was obtained with specified projectile having a striking velocity greater than that listed. Approximately 10% of the armor tested failed to meet these ballistic requirements. A greater number of failures occurred in the thinner gauges than in the heavier thicknesses. It may be noted that the ballistic requirements for the Ro 601 type armor were severe. These data were reported by the Japanese from memory and may not be accurate.

2. Army Tests

Table V lists the conditions of test for army aircraft armor.

Thickness Projectile* 0bl1-Striking Spec. quity (in.) (AP) Velocity (ft/sec) .079 7.7mm 0 1312 2-704 1476 .118 0 0 1640 .158 Homo-.197 ** 0 1960 genecus ** .236 0 2296 77 0 2362 2-704 .315 13mm Ô 1935 2-707 .355 0 1984 .394 ** 0 2034 Homogeneous. .433 ** 0 2083 0 21,32 .472 and Carburized 0 2230 .550 .591 0 2280

Table V

.632

The Japanese Army ballistic acceptance procedure differed from the Navy method only in interpertation of a failing impact. A plate failed the ballistic test under the conditions of test listed in Table V when either of the following occured:

a. A through hole in the plate, providing the striking velocity of the projectile was equal to or less than that listed.

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 b. Excessive cracking or breaking of the plate as defined by the inspector.

The Japanese Army considered acceptance testing of 10mm, to 16mm, carburized armor with 20mm A. P. projectiles, but never put it into effect.

E. WELDING CF AIRCRAFT ARMOR

In fabricating and installing aircraft armor, the Japanese employed no welding nor did they investigate the possibility or study the effects of welding of aircraft armor.

F. DEVELOPMENT OF ARMOR

In general the Japanese were thorough in their investigation of experimental armor. Enclosure (A) is a report on a typical investigation. Records indicate that the Japanese conducted complete metallurgical investigations, in addition to ballistic tests, on all experimental armor.

^{*}Frejectiles apparently identical with those described in Table IV.

The Japanese Navy and Army conducted the following experiments during the war in an attempt to improve the quality of aircraft armor:

- 1. Determination of the optimum depth of carburization for armor versus the 13mm A.F. Projectile. This experiment showed a 30% case to be the most effective against the 13mm A.F. projectile, however, a 20% depth of case was adopted for production armor because of manufacturing difficulties associated with making a heavier case.
- 2. Development of substitutional steels without nickel and molybdenum for use in armor.
- 3. Determination of the optimum heat treatment for each type armor.
- 4. Experimentation with composite armor produced by casting one steel into a mold, removing a partition, and quickly casting the second steel: This experiment was conducted in order to make armor with a face of greater depth and a higher and more uniform hardness pattern. In addition, composite armor was carburized. The results of the investigation ion, composite armor was carburized. The results of the investigation showed that a good bond could be obtained between the two steels and that carburized composite armor offered greater penetration resistance to 13mm A.P. and 25mm A.P. projectiles than did ordinary cemented armor. Ecwever, composite armor of this type apparently never was used in service. Complete details of one such experiment may be found in the Japanese accument entitled "Manufacture and Ballistic Test of Composite Armor" which has been forwarded to the Washington Document Center, under NavTechJap Document No. ND50-3157, (See Enclosure (E)).
- 5. Development of an aircraft armor for protection against high exploive projectiles. On the basis of this experiment the Japanese decided upon the use of: (1) a carbon steel of approximately S.A.E. 1040 composition, and (2) a silicon-manganese-chromium steel. A straight quench and temper to produce a hardness of 250 to 400 BHN was used. Table VI lists the average limits obtained with this armor under high explosive projectile fire.

Table VI STRIKING VELOCITY (ft/sec)*

| Thickness | Aver | age Plate Limit Ver | sus: |
|--------------------------------------|----------------------|------------------------------|--------------------------------------|
| of Armor (in.) | 20mm HE | 30mm HE | 40mm HE |
| .632 .550 .472 .394 .315 | 2788 2296 1804 | 2296 2066 1837 1640 | 2427 2165 1902 1640 1378 |

*All projectiles were fuzed with a nose detonating fuze of the air column type (no firing pin).

A limit was an estimate of the minimum striking velocity of a projectile required to cause:

- a. A depression in the plate of 15mm depth or
- b. Excessive cracking of the plate, as decided by the firing official.
- 6. Ballistic tests of verious combinations of spaced aircraft armor.

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This experiment was conducted in an attempt to determine if two or more thin sections of armor separated by air were equivalent in protection afforded to one solid section of armor of equal weight. Some combinations of armor spaced approximately 15mm apart were found to afford approximately 20% more protection than an equal weight of solid armor.

PART II - TANK ARMCR

The Japanese Army used both homogeneous and cerburized armor for tanks. The analysis of the homogeneous armor was C:C.25-0.35; Mn:0.60; P:0.030; S:030; Si:0.35; Ni:2.0-4.0; Cr:0.8-1.8; and Mo:0.20-0.50 with 0.7 to 1.5% tungsten sometimes substituted for the molybdenum. The armor was quenched and tempered to a herdness of approximately 320 BHN. Table VII lists the conditions of test for acceptance.

Table VII
HOMOGENEOUS TANK ARMOR PENETRATION TESTS

| Projectile | Thickness of Armor (inches) | Cbliquity | Striking Velocity (ft/sec) |
|------------|--------------------------------------|-----------------------------|--------------------------------------|
| 7.7mm Ball | .079 .118 .158 .197 | 0° 0° 0° | 1210 1570 1890 2230 |
| 7.7mm A P | .236 .314 .394 .473 .550 | 30° 0° 0° 0° 0° | 2230 1640 2000 2220 2580 |
| 20mm A P | .630 .787 | 30° 0° | 1470 1150 |
| 37mm A P | .983 1.18 1.38 1.57 | 30° 0° 0° | 1310 1180 1380 1570 |
| 47mm A P | 1.97 | 00 | 1570 |
| 37mm A P | .236 to .787 | 00 | 1080 |
| 47mm A P | .983 to 1.57 | 00 | 1150 |

In general, carburized armor for tanks was a nickel-chrcme-molybdenum steel of the same composition as the homogeneous armor. However, a nickel-chrcmium-molybdenum steel with approximately 0.48% carbon also was used for face-hardened armor. Table VIII gives the tallistic requirements for the above armors.

Table VIII

BALLISTIC REQUIREMENTS FOR TEST ARMOR

| Type of Armor | Thickness (inches) | Projectile | Obliquity (°) | Striking Velocity (ft/sec) |
|--|--|-------------------------------------|----------------------------|--|
| Ni-Cr-Mo 0.24-0.28C Face Fardened | .236 .314 .394 .473 .787 .983 1.18 1.38 1.57 | 7.7mm A P " " 20mm A P 37mm A P " " | 0000000 | 2156 2320 2490 2650 1800 1640 1830 2000 2150 |
| Sheck Test | .236- .787 | 37mm A P | 0 | 1080 |
| | .983- 1.38 | 47mm A F | 0 | 1310 |
| Ni-Cr-No 0.48C Face Hardened | .158 .197 .236 .314 .394 .473 | 7.7mm A P | 0 0 0 0 0 0 | 1640 1960 2296 2450 2610 2710 |

PART III. LANDING BARGE AND GUNSHIELD ARMOR

During the war, the Japanese Navy used very little steel in landing barges and gunshields which could properly be called armor. Instead of armor steel, they used mild steel and a basic open hearth semi-kilned structural type steel having a tensile strength of 80,000 to 90,000 FSI. The average analysis of the structural type steel, called "Ducol" by the Japanese, was C:0.24,Mn:1.4, Si: 0.35, S:0.035, F:0.035, Ni:trace and Cu:0.20. The high manganese content gave increased strength to the steel and the corpor addition supplied corrosion resistance. Nevertheless, this steel could not be classified as armor nor did the Japanese consider it armor, as it was not subjected to ballistic acceptance tests. "Ducol" and mild steel were used for gunshields on mounts up to the 12.7mm anti-aircraft type. The side, back, and top of these gunshields usually were constructed of one or two thicknesses of 3.2mm plate. The front side of the gun shield was made of 8mm or 12mm plate.

Frior to the wer, the Japanese Navy experimented with and used many types of armor in gunshields. A document entitled "Firing Tests of Non-Nickel Shields", which has been forwarded to the Washington Document Center under NavTechJap Document No. ND50-3166 (see Enclosure (E)), lists various steels for gunshields and gives their respective ballistic performance as well as a good description of experimental ballistic testing methods employed by the Japanese Navy. Table IX, lists the steels investigated in the experiment and gives the mechanical properties.

Table X lists the ballistic results obtained on these types of steel.

An austenitic manganese steel with a small amount of nickel and chromium was used as armor by the Japanese Navy. This armor generally was less than $1\frac{1}{2}$ inches in thickness and since it was non-magnetic, it was used for protection when the operation of nearby instruments would be hampered by ferromagnetic armor.

Table IX

| | | .+ m | N N | ~~ | ~ | | | اه |
|-----------------------|--------------------------|--------------|------------------------|------------------------|-----------|-------------|---------|-------------------|
| | BHN | 254 253 | 88 | 887 | 265 | | | 28 8 8 8 |
| rties | Izod Value (ft. 1bs.) | 51.2-56.4 | 37.3-39.6 35.6-38.3 | 32.7-38.7 25.2-28.3 | 63.5-60.5 | 230 | >30 | Charpy 5.5-9.9 |
| al Prope | Elong. | 24.1 | 20.8 19.9 | 21.5 | 24.0 | ^ 18 | >19 | >20 |
| Mechanical Properties | T.S. (ps1) | 120,000 | 134,000 | 135,000 136,000 | 130,000 | > 100,000 | 110,000 | 85,000 93,000 |
| | Direction | 나는 | ㅂ | HH | ş | | | |
| | ກວ | 0,28 | 0,27 | 0,27 | 0.14 | | 0.90 | 0.044 |
| | Mo | %.0 | 0.26 0.27 | 0.24 0.27 | 0.54 0.14 | | | |
| sis (% | G. | 1,35 | 1.36 | 1.37 | 0.95 | 1.8- 2.2 | 0.80 | |
| . Analy | N1 | 0.27 | 0.25 | 0.27 | 4.12 | 3.7- | 3.0 | ** |
| Chemical Analysis (%) | S. | 0.17 | 0.13 | 0.14 | 0.21 | 0.05 | 0.05- | 0,085 |
| 0 | 덃 | 1,31 | 1,30 | 1,30 | 0.73 | 0.30- | 0.30- | 1.20 |
| ,,, | U | 0,38 | 0.35 | 0,34 | 0.28 | 0.43- | 0.38- | 0.27 |
| Type of | Armor | 25mm BK 0.38 | 18mm BK | 12mm BK | NG | NVNC | CNC | DS |

Table X
BALLISTIC RESULTS WITH TEST STEELS

| Type of Armor | Thickness (mm) | Hardness (BHN) | Frojectile | Obliquity (O) | Limit* Velocity (ft/sec) |
|---|--|--|------------|--|--|
| BK BK BK NC NVNC DA CNC DS DA BK BK DS DA | 25 25 18 18 25 25 25 18 20 20 12 12 12 | 254 254 282 282 265 250 288 288 | 20mm A P | 0 30 0 30 0 0 0 0 30 0 30 0 | 1700 1820 1350 1748 1680 1638 1472 1305 1250 1712 1530 1810 1270 1630 |

*Limit velocity was estimated from firing data and then corrected by Demarre formula for varietions in gauge. (For further details see NavTechJap Report "Japanese Heavy Armor", Index No. 0-16.)

PART IV. CTHER TYPES OF LIGHT PROTECTIVE MATERIALS

For protection against the effects of gun fire, the Japanese apparently used, in addition to steel, only bullet-resistant glass and sponge rubber. Bullet-resisting glass was used for aircraft windshields. It was manufactured in various thickness up to three inches and consisted of several laminations of heat-treated glass separated by layers of polyvinyl formaldehyde resin. This glass was designed to provide protection against 7.7mm and 13mm ammunition. Details of experiments performed by the Japanese with bullet-proof glass may be found in pamplets forwarded to the Washington Document Center under NavTech Jap Document Nos. ND50-3147 and ND50-3189 (see Enclosure (E)). Natural rubber between layers of vulcanized rubber was used for self-sealing protection of fuel cells in aircraft.

The Japanese used no materials such as nylon, silk, leather, or glass cloth for the protection of personnel or equipment against the effects of gunfire. With the exception of a few experimental ballistic tests on very thin aluminum plate, there was nothing to indicate that the Japanese were investigating aluminum or magnesium alloys for use as armor.

FART V. DISCUSSION

Enclosure (D) lists Japanese light armor samples shipped to the Naval Frowing Ground, Dahlgren, Virginia. Pertinent documents recovered in JAFAN are listed in enclosure (E) and have been forwarded to the Washington Document Center. Experiments performed at the Tohoku Imperial University, SENDAI, are listed as NavTechJap Documents No. ND50-3146, ND50-3162, and ND50-3164. Enclosure (B) is a discussion of the static torsion test performed at the University.

Date available indicate that Japanese Light armor steel (homogeneous and carburized) used in aircraft, landing barges, and gunshields was inferior to U.S. Navy armor steel for protection against the effects of gunfire.

While their homogeneous armor appeared to the Japanese to possess the correct hardness for supplying protection, this armor, at a hardness of 500 to 630 BHN, offered poor penetration resistance to overmatching projectiles which were used against it. In addition, the shock properties of the armor no doubt were very poor. However, Japanese data indicated the hardness of the armor to

0-36-1

be optimum.

A recent report entitled "Experimental Mn-Va Steel Torsicn Test" which has been forwarded to the Washington Document Center as NavTechJap Document No. ND50-3162 (see Enclosure (E)), shows that the calculated thickness of armor required to stop a projectile at a specified velocity was less at lower tempering-temperatures than at higher ones. In other words, it was evident to the Japanese that as the hardness of armor increased, the thickness required to afford a given amount of protection decreased.

The carburized armor manufactured by the Japanese generally was of good quality. However, it was noted that no improvement in the quality of this armor was effected during the war. This may be attributed to the substitution of other alloying elements for nickel and molybdenum.

There appeared to be some reluctance on the part of the Japanese Navy to use aircraft armor. The Army believed aircraft armor necessary and did more development work on it.

ENCLOSURE (A)

EXPERIMENTS WITH SPECIAL ARMOR PLATE, FOR USE IN PLANES, MANUFACTURED BY THE DAIDO STEEL CO. LTD.

(Translation)

Report No. 03104

Material Report 0330

Period of Experiment:

10 April 1942 - 20 December 1942

Experiment No:

No. 103 Material Test 17

Type of Notification Bulletin:

Air Headquarters Top Secret Bulletin #11115 on 20 Dec., 1940 (See Addendum)

Persons Compiling Report:

Navy Senior Technician Hideo YAMAGUCHI Navy Lieut. (Technical) Keishi MATSUBARA

Purpose:

To test the special armor plate for use in planes manufactured by the Daido Steel Co., Ltd. and to determine the value of using protective armor plate on planes.

RESTRICTED

ENCLOSURE (A)

INTRODUCTION

A. OUTLINE OF RESULTS

1. Firing Test

- a. The bullet resistance power of BK3 is slightly superior to BKg, but it is difficult to say that it is perfected when judged from the standard necessary for armor plate.
- b. Both BK3 and BKg generally are brittle. Compared to the "Ro 601" used at present, they have a great tendency to begin to crack and peel. This is especially true of BKg.

2. Test for Composition of Materials

- a. BK3 is made up of chrome, manganese, molybdenum steel (carbon content 0.3). BKg is made up of silicon, manganese, chrome steel (carbon content 0.37).
- b. The degree of hardness of BK $_3$ and BK $_3$ (inner degree of hardness of cemented plates) is 530 Hv and 600 Hv respectively.
- c. Even though the cemented thickness averages 15%, the degree of hardness of cemented parts of both steels is very low and no effect of cementation is noticed in 2, 4mm plates of BK3.
- d. There are many non-metallic foreign inclusions in both types of armor plate and there is marked bending and breaking.
- e. The degree of crystallization of BK3 is G6, of BKg is G7.

3. Opinions of OinC of Navy Air Department

- a. Among the materials tested, BK3 is a non-nickel steel plate and so it is not quite suitable, but if its flexibility were improved, there would be nothing to prevent its use.
- b. Since BKg is very brittle, it is thought that it has few useful characteristics.

4. Opinions of OinC of Navy Technical Dept. (Air)

(TN: none given)

B. PERSONS IN CHARGE OF EXPERIMENTS AND ASSISTANTS

1. Material Supply Dept:

Fsunahiko YOTEI, Navy Senior Technician Hideo YAMAGUCHI, Navy Senior Technician Keishi MATSUBARA, Navy Lieut. (Technical) Masao SEMBA, Navy Technician

2. Equipment Department:

Heishiro NISHIDA, Navy Capt.
Masa KUGIMIYA, Navy Lieut.
Akio YANAKA, Navy Lieut. (Technical)
Yukishige MATSUMOTO, Navy Technician

CHAPTER I. PURPOSE

To test special armor plate, for use in planes, manufactured by the Daido Steel Co. Ltd, and to determine the value of using protective armor plate on planes.

CHAPTER II. RESULTS AND OPINIONS

A. RESULTS

1. Firing Tests

- a. The critical angle in oblique-firing tests and bullet-resistive powers in direct-firing-tests are shown in Table I(A). The bullet-resistive power of BK3 is slightly superior to BK8, but it is difficult to say that it is perfected when judged from the standard necessary for armor plate.
- b. Both BK3 and BK8, generally, are brittle, and compared to "Ro 601" used at present, they have a great tendency to begin to crack and peel. This is especially true of BK8.

2. Test for Composition of Materials

- a. BK3 is made up of chrome, manganese, molybdenum steel (carbon content 0.37) (See Tables II(A) and III(A).
- b. The degrees of hardness of BK $_3$ and BK $_8$ (inner degrees of hardness of cemented plates) are 530 HV and 600 HV respectively. The results of the tempering tests are the same.
- c. Even though the cemented thickness averages 15%, the degrees of hardness of cemented parts of both steels is very low and no effect of cementation is noticed in 2 and 4mm plates of BK3.
 - d. There are many non-metallic foreign inclusions in both types of armor plate and there is marked bending and breaking.
 - e. The degree of crystallization of BK3 is G6 (Gakushinho method of testing) and of BK8 is G7.

B. OPINIONS

- 1. Since the same constituent elements have been used in both the cemented and non-cemented plates, they are considered unsatisfactory. It is thought that one must choose constituent elements (particularly carbon content) which are most suitable to each type of plate respectively.
- 2. The bullet-resistive strength of non-cemented BK3 is roughly the same as "Ro 601" now being used, but it is less flexible. Cemented BK3 plate is on the whole, as good, but on testing its constituent elements, inner hardness, thickness of cementation, etc, it was found that these characteristics (flexibility, most of all) had to be increased.
- 3. Non-cemented BKg plate does not come up to the bullet-resistive strength of the presently used "Ro 601" and is extremely brittle. BKg cemented plate also is brittle. It is necessary again to check on the constituent elements, inner hardness, etc. and follow a method whereby the carbon content is reduced to 0.25-0.30.

4. The characteristics of BK3 are not satisfactory, but since it is a non-nickel plate, there is a good chance that it can be used at the present time. (However, it is thought that further research concerning the improvement of its flexibility is necessary.)

If no further steps are taken to improve BKg, it is thought that it has little practical value.

5. Because there are foreign objects in the plates and there is bending and breaking, further improvements are necessary.

CHAPTER III. DEVELOPMENT OF TESTS AND RESULTS

A. TEST MATERIALS

The tested materials were two kinds of special steel plates, BK3 and BK8, to be used as armor plate and manufactured experimentally by the Daido Steel Co., Ltd. Table IV(A) lists the measurements and number of samples provided by the Company. The records of the plates are given in Tables II(A) and III(A).

B. FIRING TESTS

- 1. Vickers type 7.7mm fixed MG, Model 3 Modification 1; 13mm fixed MG used on test planes.
- 2. Direct-Firing Test: Ordinary bullets and armor piercing shells, 7.7mm, were fired at a range of 25 meters. Details of results of the test for bullet-resistive strength under direct fire are shown in Table V(A) and Figure 1(A).

The bullet-resistive strength of BK3 non-cemented plates toward ordinary bullets surpassed the present standard used for bullet-resistive strength, but against armor-piercing shells, did not come up to the present standard. The bullet-resistive strength of BK8 non-cemented plates toward both ordinary and armor piercing shells does not satisfy the present standard used for bullet-resistive strength.

The cemented plates, except for BKg μ mm plate, come up to the standards now used, but the bullet-resistive strength of the 2 and μ mm plates is the same as the non-cemented plates. An increase in resistive strength due to cementation is seen for the first time in the 6mm plates. This is noted in the article about the material test. The cementation thickness in the 2 and μ mm plates is very thin and it is thought that there is very little increase in surface hardness by cementation. Also, when comparing the resistive strength of BK3 and BK8, the strength of BK3 slightly surpasses that of the latter. Generally speaking, it is difficult to say that the resistive strength of either is good.

Both BK3 and BKg generally are brittle. Compared to "Ro 601" used at present, they have a great tendency to begin to crack and peel. This is especially true of BKg. This insufficient flexibility is thought to be due to the composition of the materials. Moreover, the thin plates, especially 2 and 4mm, are generally brittle. When one observes these cracks, it is obvious that they are in the direction of the rolling process of the plate.

3. Oblique-Firing Test

a. Using the same plates as in the direct-firing tests, the firing angle was changed by various degrees. The results which tested the

critical angle of oblique firing are given in Table VI(A) and Figure 2(A). The balanced firing angle of the 7.7mm armor piercing shell fired at 100 meters range toward the various plates is given in Table VII(A).

By increasing the firing angle toward both BK₃ and BK₈, their resistive strength increase is almost the same. Under conditions of a firing range of 100 meters and at an identical firing angle, both cemented and non-cemented plates fall within these limits: plates 2 and 4mm thick, $50^{\circ}-60^{\circ}$; plates 6mm thick, $25^{\circ}-35^{\circ}$. Also, because of the firing angle toward both BK₃ and BK₈, there was the same tendency to split and peel as when firing directly. They were somewhat brittle, especially BK₈.

b. Firing test by 13mm MG: Results of the investigation of conditions of damaged armor plate from a 13mm MG armor piercing shell with the normal service charge used in aircraft, are shown in Table VIII(A). As the thickness of the plate increased, the diameter of the hole increased. There were no other particular changes.

Figure 3(A) and Figure 4(A) show conditions of the plates following firing.

4. Materials Tested

- a. Appearance and measurements: There was a thin oxidized layer over the outside layer of the plates provided for the test. The materials were bent very little, and there were no noticeable defects, such as folds, resulting from the rolling process. All the 2 and 4mm plates had variations and were not good. The results of investigating the differences in thickness of the plates are given in Table IX(A). Except for BK3, 2mm plates (cemented) and 6mm plates (cemented) all were good. (Table IX(A) is on page 29.)
- b. Range of hardness of samples provided: The measurement of hardness of material provided is given in Table $X(\Lambda)$. The average hardness of BK3 is 50-52 HRC. and of BK6 is 53-55 HRC. Although BK8 is slightly harder, each plate's hardness range is about the same. The surface hardness of cemented plates is generally low. Particularly in the care of the 2 and 4mm plates, the cemented surfaces are nearly as hard as the rear surface. The effect of cementation is very slight.
- c. Range of cross-sectional hardness: The results of the investigation of cross-sectional hardness for each cemented plate are shown by curves in Figure 5(A). The thickness of cementation is shown in Table XI(A). The hardness of the cemented portions generally are low, and BK3 2 and 4mm plates, especially, show no effect from cementation. BK8 2 and 4mm plates show few results from cementation. The hardness of the inner part of BK3 is about 530 Hv and of BK8 is about 600 Hv (C-4-2, however, is slightly lower). The hardness of BK8 is high.
- d. Non-metallic foreign inclusions: On investigating the foreign inclusions according to the method set forth by the Gakujitsu Shin-kokai (Scientific Advancement Society), exceedingly many were found in all plates, especially many traces of oxidized substances. BK8, as compared to BK3, had many more non-metallic foreign inclusions. This condition is shown in Table XII(A) and Figure 6(A).

- e. Microscopic formations: On investigating for miscroscopic formations of each plate which had been provided, all were found to have similar Martinsite formation. There were no other particular differences (see Figure 7(A)). The plates and their formations, which had been annealed at 900° C for 1 hour, are shown in Figure 8(A). Bending and breaking of each plate was severe and examples of this are shown in Figure 9(A).
- f. Degree of crystallization: The results of measuring the degree of crystallization have been obtained by the cemented method set up by the Gakujitsu Shinkokai and are shown in Table XIII(A). The crystallization of BK3 is G6 and of BK8 is G7. BK8's crystals are smaller than those of BK3 and are almost uniform in size. Figure 10(A) shows this condition.
- g. Hardness, tempering relationship: Selecting a plate to be tested from among the non-cemented plates, they obtained a plot showing the hardening and tempered hardness of BK3 to be about 5 $\rm H_{RC}$ lower than that of BK8.
- 6 April 1943.

ADDENDUM

REFERENCES FOR TESTS

Air Ho. Top Secret Bulletin No. 11115.

20 December 1940

From: To

OinC, Naval Air Headquarters, Administration Dept.

OinC, Naval Air Technical Depot, Administration Dept.

Subject:

References concerning tests of special steel plates for the use of aircraft manufactured by the Daido Steel Co., Ltd.

We will send samples of the above plate to you and request that upon testing it, you send us your results and opinions by the end of March. 1941.

The expenses involved will be paid by the Military budget from the Ship Building Ordnance and Repair Fund, Ordnance Fund, and Air Technical Experiment Fund.

METHOD OF PROCEDURE

Air Technical Depot Top Secret Order No. 148.

The procedure to be carried out in the experiments with special steel plate for aircraft manufactured by the Daido Steel Co., Ltd. referred to in Air Headquarters Top Secret Bulletin No. 11115, 1940, is as follows:

The OinC of the Material Department is to be in charge of these experiments.

10 January 1941

C. O. Naval Air Tech. Depot - Misao WADA

Test these plates and determine their practical value as armor plate for airplanes.

Nature of Research 2.

- Firing tests. a.
- b. Tests of uniform characteristics of materials.
- Physical and mechanical characteristics.

Date and Place 3.

- Date: 10 January 1941 to end of March 1941.
 Place: Material Supply Dept., No. 4 Laboratory, Equipment Dept. No.1. b.

Method of Procedure 4.

Test materials а.

(1) BK3 (thickness 2 and 4mm), 2 plates each, total 4 plates.
(2) BK8 (thickness 2, 4, and 6mm), 2 plates each, total 6 plates.

Firing tests: Fire 7.7mm ordinary and armor piercing shells. At a firing range of 25 meters, measure the initial velocity. In addition to measuring the bullet-resistive strength at direct fire, measure the acute angle under conditions of oblique firing if there is room left on the steel plate.

Follow, generally, the "Special Steel Plate Testing Regulations" given in Air Headquarters Top Secret Bulletin No. 4379, 1940, in carrying out these experiments.

Test for uniform characteristics of materials

Examine outer appearance

Investigate range of hardness

- (2) (3) Investigate differences in plate thickness
- Make a reduction test (TN: for constituent elements)

Other tests thought necessary

Determine physical and mechanical characteristics

- Heating procedure for plates tested will be decided upon speially (TN: In another bulletin?)
- Items to be measured
 - Measure unusual points (a)
 - Test qualities caused by heat process
 - Test for microscopic formations

Test for other things thought necessary

Work Schedule 5.

Make all tests and compile reports from 10 January 1941 to end of March 1941.

6. Report

Submit a report as soon as possible after the tests are made.

Persons in Charge of Tests

Material Supply Dept:

Koi KAWAMURA, Navy Comdr. (Engineer) Kazuma KOTO, Navy Lt. Comdr. (Ordnance) Mitsugu IKOMA, Navy Ensign (Ordnance)

Equipment Dept:

Chosu SHINOZAKI, Navy Comdr. (Ordnance) Tsao KUBO, Navy Lt. (Ordnance)

Table I(A)
RESULTS OF FIRING TESTS

| | | HEDOETO | or Friding India | | | | | | |
|---|-----------------|--|--|--|--|----------------------------------|--|--|--|
| Projectile | Plate | Thick- ness | Direct F | iring | Oblique | e Firing | | | |
| 7 · · · · · · · · · · · · · · · · · · · | · . | (mm) | Impact Speed (m/sec) | Range (m) | Impact Speed (m/sec) | Angle (°) (Range - 100 m) | | | |
| 7.7mm Ball | BK3 | 2.07 2.55 3.98 4.23 6.03 6.28 | 475 497 636 617 >722 >729 | 532 479 174 216 > 31 > 12 | | | | | |
| | BK ₈ | 2.25 2.17 4.06 3.87 6.24 6.04 | 425 345 539 545 678 > 729 | 658 923 385 373 93 > 22 | | : | | | |
| 7.7mm A.P. | BK3 | 2.05 2.50 4.00 4.23 6.04 6.15 | 345 344 380 405 560 630 | 923 936 796 715 340 187 | 680 680 680 680 680 680 | 60 58 54 50 14 18 | | | |
| 3 | BKg | 2.25 2.17 4.01 3.94 6.22 6.02 | 333 >337 372 355 532 583 | 995 820 888 401 290 | 680 680 680 680 | 47 55 29 20 | | | |

Note: Treatment - Carburized

TABLE OF RECORDS OF SAMPLE STEEL PLATES, DAIDO STEEL COMPANY, (Ltd.)

| | | | ' | | | | Ind L | i | | |
|---------------|--|-----------------------------|--|------------|--|------------|---|---------------------|--|--|
| | | Mo | 0 | | | | ick a | | | |
| | | ප් | 67 - | ì | 5 | 3 | nm thi | <u>;</u> | | |
| | | ਲ | 0,00 | | 0.00 | 72.0 | BK3 10 | | | |
| | ; (%) | ß | 030 | | 0.031 | 1000 | irs of | 3 | | |
| | Chemical Elements $(\%)$ | Д | 010 | 0.01/ | 660 0 | 0.022 | sting ba | | | |
| | ical E | щ | 7. | - · - | 50 | 706 | Shee | ses. | | |
| 18 Karch 1942 | Cher | Si | 75 0 | 2.0 | 6 | 3 | Method of Cementation: Sheeting bars of BK3 lOnm thick and BK8 20mm thick are cemented 8 hours at 900° C and then rolled to various thicknesses. | | | |
| | | C | 15.0 | 10.0 | 20 | 75.0 | Cement | ious th | | |
| | | Materials | 35 0 97 1 66 0 0 0 0 0 0 21 1 35 0 15 0 0 0 m2 m2 m2 | ONI TOLINA | 20 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 | 31-hill-or | *Method of Cementation: Sheeting bars of BK3 10nm thick and BK. 30nm + high one comented & hours at 90000 and then rell- | ed to var | | |
| | tes** | 6mm Thick | 2 | 2 | 2 | 2 | 8 | | | |
| | of Pla | Znm 4mn Thick Thick | 2 | 2 | 2 | 2 | 8 | | | |
| | Number | Number Znm Thick 2 | | | 2 | 7 | 8 | | | |
| | Cemented* | | Yes | No | Yes | No | tes | P. | | |
| | Type Smelting Cemented* Number of Plates** | reconn | T 115269 | | T 36503 | | Total Number Plates | ***All 450mm square | | |
| | Type | | БЖ3 | | BKg | | Total | ***A11 | | |

Table III(A) CHARACTERISTICS OF MATERIALS

| Kind of Plate | Thick- ness | Test | Plate No. | | Annealing | - • | Tempering Hardness | Cemented |
|------------------|----------------|----------|--------------|---------------------------|---------------------------------|---------------------------------|--------------------------------|----------|
| P. | (mm) | | | Normal- izing Temp. | Harden- ing Temp (in oil) | Temper- ing Temp. (6 hrs) | (H _{RC}) | |
| BK ₃ | 2 | Т 115269 | 2-1 2-2 | | 870° 870° | 180° 180° | 50 . 8 50 . 3 | No No |
| | | p . | C-2-1 | 900°C AC* | 870° | 200° | front 54.3 back 49.8 | Yes |
| | | | C-2-2 | 900° AC | έ70° | 200° | front 53.0 back 49.0 | Yes |
| | <i>L</i> , | т 115269 | 4-1 4-2 | | . 870° 8 7 0° | 150° 150° | 50.8 49.8 | No No |
| - - | 1 , | | C-4-1 | 900° AC | 870° | 200° | front 53.0 back 45.5 | Yes |
| | | | C-4-2 | 900° AC | 8 7 0° | 200° | front 55.0 face 49.0 | Yes |
| 1 | 6 | т 115269 | 6-1 6-2 | | 870° 870° | 120° 120° | 50.8 50.3 | No |
| | i i | | C-6-1 | 900° AC | 870° | 200° | front 55.5 face 47.5 | Yes |
| | | | C-6-2 | 900 ⁰ AC | 870 ⁰ | 200° | front 56.0 face 51.0 | Yes |
| вка | 2 | т 36503 | 2-1 2-2 | - | 860° 860° | 180° 180° | 53.5 54.3 | No No |
| | | 8 | C-2-1 | 900° AC | 860° | 200° | front 53.3 face 52.5 | Yes |
| | | | C-2-2 | 900° AC | 860° | 200° | front 54.3 face 52.7 | Yes |
| | 4 | т 36503 | 4-1 4-2 | | 860° 860° | 150° 150° | , 50.8 49.8 | No No |
| | | | C-4-1 | 900° AC | 8600 | 200° | front 55.8 face 52.0 | Yes |
| <u>-</u> . | | | C-4-2 | 900° AC | 860° | - 200° | front 54.5 face 50.8 | Yes |
| | 6 | т 36503 | 6-1 6-2 | | 860° | 120° 120° | 53.0 53.5 | No No |
| | | | C-6-1 | 900° AC | 860° | 2000 | front 56.0 face 49.8 | Yes |
| | | | C-6-2 | 900° AC | 8600 | 200° | front 57.3 face 52.0 | Yes |

*AC - air cooled

Table IV(A)
THICKNESS AND NO. OF PLATES

| Туре | C | emented | - | Non | -Cement | ed | Total |
|------|------|---------|-----|-----|---------|------|-------|
| | 2mra | 4mm | 6mm | 2mm | 4mm | 6ram | |
| вк3 | 2 | 2 | 2 | 2 | 2 | 2 | 12 |
| BKg | 2 | 2 | 2 | 2 | 2 | 2 | 12 |

Table IX(A)
VARIATION IN THICKNESS OF TEST PLATES

| Туре | Treatment | Variatio | on in Thickne | ss (mm) | |
|-----------------|------------|-----------|---------------|-----------|--|
| | | 2mm Plate | 4mm Plate | 6mm Plate | |
| DV | | 1.95-2.19 | 3.70-4.10 | 5.90-6.20 | |
| BK ₃ | Carburized | 2.40-2.95 | 4.00-4.27 | 6.00-6.90 | |
| DV | | 2.15-2.30 | 4.00-4.14 | 6.07-6.30 | |
| BKg | Carburized | 2.10-2.27 | 3.80-4.10 | 5.88-6.10 | |

ENCLOSURE (A), continued

Table V(A)
DAIDO STEEL PLATE - RESULTS OF EXPERIENTS ON PROJECTILE RESISTANCE

| | | i |
|------------------|---|---|
| | | ĕ |
| | L Depot | Type 3 |
| | chnical | (Air) |
| | Ę | 9 |
| | Air | ixed |
| 1 | Depot, | 7mm, F |
| | Branch | Type 7. |
| 74/5T | Mace: Firing Range, Branch Depot, Air Technical Depot | Vickers Type 7.7mm, Fixed MG (Air) Type 3 Mod 1 |
| 5 May | Firing | Gun: |
| ate: 15 May 1942 | lace: | fachine Gun: |

| Type | Target | et | Type Shell* | Firing | Corre- | Pene- | Diam- | Depth | Penetration Condition |
|---------|-----------------|---|----------------|--------------------------------------|--|--------------------------------------|---------------------|--------------------------|--|
| | No. | Thick- ness (mm) | | (正/sec) | ing Range (m) | | of Hole (mm) | Inden- tation (mm) | |
| BK3 | 7- 1: | 2.19 2.19 2.13 2.00 | Соппол | 384 424 448 444 448 | 782 598 662 560 | No No No Yes | | 43 | 2 cracks - 70, 70mm 2 cracks - 80, 40mm 3 cracks - 60, 75, 35mm 3 cracks - 60, 40, 40mm |
| | | 22.22.23 22.22.23 23.24.28 23.24.28 23.24.28 23.24.28 23.24.28 23.24.28 23.24.28 23.24.28 23.24.28 23.24.28 23.24.28 23.24.28 23.24.28 23.24.28 26.28 | A.P. | 333 | 9774 990 780 986 825 798 995 | No Head Yes No Yes No | 5x5 8x8 10x20 | | Projectiles head only 4 cracks: 20, 25, 20, 20mm |
| | - 2 | 2.01 2.02 2.01 2.02 | Common | 17.7 17.7 18.5 25.4 25.4 | 689 612 507 420 | No No No Yes | 8x8 | | l crack: 130mm l crack: 110mm l crack: 90mm |
| | | 2.04 2.03 2.03 | A.P. | 343 | 97% 27% 300 | No Part Part | 4x4 · | 0.5 | l crack: 70mm Penetrated armour, stopped, caused crack 75mm long |
| | \$* * | 2.50 2.45 2.49 2.48 | Соптоп | 528 395 459 472 | 410 746 572 541 | Yes No No Yes | 30x20 20x55 | 2.2 | 2 cracks - 40, 40mm 2 cracks - 70, 30mm |
| | | 2,50 2,45 2,54 | A.P. | 344 391 387 | 936 758 772 | Head Yes Yes | 10x4 15x6 8x8 | | Head only penetrated Projectile core only |
| | C-2-2 ** | 2.97 2.51 2.48 | Сонипоп | 687 760 783 | 346 497 514 | Yes No No | 25x1.7 | | 1 crack 30mm 1 crack 50mm |
| #Reduce | "Reduced charge | | | | | | | | 444Includes blister plate |

"Reduced charge

**Includes blister plate

ENCLOSURE (A), continued

| X | Penetration Condition | 22 | 2 cracks - 60, 100mm 2 cracks - 20, 20mm 1 crack - 25mm | 3 cracks 40, 50, 60mm | | The steel plate was torn away | 1 crack - 40mm 1 crack - 20mm 1 crack - 60mm | Plate torn away (55x/Umm) | | |
|---|-----------------------|-------------------------------|---|-----------------------|------------|-------------------------------|--|---------------------------|--------|------------------|
| : RESISTANC | Depth | Inden- tation (mm) | | | | | | | | |
| PROJECTILE | Diam- | of Hole (mm) | 40x18 30x27 30x25 30x25 35x18 | 5x5 20x30 25x25 | V. | | 20x20 20x20 15x15 30x25 | 15x15 15x15 20x20 | 20x55 | 20x25 |
| ntinued) MENTS ON | Pene- | 92 = 143 | No Yes Yes Yes No | Head Yes Yes | No No | Yes | Yes Yes Yes No No | Yes Yes Yes | Yes | Yes |
| Table V(A)(Continued) RESULTS OF EXPERIMENTS ON PROJECTILE RESISTANCE | Corre- | spond- ing Range (m) | 1006 552 538 722 678 | 995 930 944 | 580 675 | 276 | 662 684 732 844 1012 | 974 910 796 | 976 | 11 16 |
| 1 | Firing | speed (m/sec) | 731 731 747 747 747 747 747 747 747 747 747 74 | 333 346 343 | 617 957 | 343 | 124 124 339 330 329 | 337 350 380 | 342 | 343 |
| STEEL PLATE | Type | SUBIL | Ordinary | A.P. | Common | A.P. | Conmon | A.P. | Common | A.P. |
| DAIDO | Target | Thick- ness (mm) | 222222222222222222222222222222222222222 | 2.28 2.30 2.25 | 2.23 | 2,20 | 2.2 2.2 2.2 2.2 2.2 2.2 | 2.18 2.13 2.26 | 2.17. | 2.20 |
| | | No. | 2-1 | | 2-5 | | C-2-1 ** | - | C-2-2 | |
| | Type | | BKg | | 7 | | 1. P. P. | | | |

*Reduced charge

ENCLOSURE (A), continued

Table V(A)(Continued)
DAIDO STREE PLATE - RESULTS OF EXPERIMENTS ON PROJECTILE RESISTANCE

| Hole Fatton Hole Hole | |
|--|--|
| 234 No | No.· Thick- ness (mm) |
| 704 Head 8x4 Style Style | 4-1 4.05 Common 4.05 4.03 4.03 |
| 238 No | 4.06 4.06 4.08 4.08 4.09 |
| 552 Head 6x7 4.6 Back slightly crack 986 No 7x7 2 1 crack - 15mm 5alling of 1 crack - 15mm 540,000 1 crack - 15mm 540,000 500 1 crack - 15mm 500,000 5 | 4-2 3.91 Common 3.92 3.92 |
| 120 Yes 8x9 2 cracks - 30, 20mm bas 338 No 7 3 cracks - 20, 25, 25, 229 No 120 Yes 27x35 2 cracks - 30, 25, 26 240 No No 1 | 44.64.64.64.64.64.64.64.64.64.64.64.64.6 |
| 980 No 6x6 240 No 3.7 108 Yes 22x14 94 Yes 17x3.2 200 Yes 20x10 995 No 1.4 | 2-4-1 4.20 Common 66 ** 4.15 65 4.15 65 |
| 240 No 3.7 108 Yes 22x14 94 Yes 17x3.2 200 Yes 20x10 995 No 1.4 | 4.22 A.P. 33 |
| 995 No 1.4 698 Part. | 2-4-2 4,22 Connop 6 4,4 4,24 6 4,24 6 |
| | 4.22 A.P. |

Table V(A) (Continued) DAIDC STEEL PLATE - RESULTS OF EXPERIMENT ON PROJECTILE RESISTANCE

| Penetration Condition | . a | Slight swelling at back 1 crack - 150mm Back (35x40mm) falling off 2 cracks - 100, 60mm Back (30x25) falling off, one part flew of | Scratch-like scar Scratch-like scar Back (27x31mm) falling off Slight swelling at back | 4 parts flew off 7 parts flew off | (20x25mm) falling off Slight swelling | 90mm Vert crack front 60mm crack | 230mm crack from left to right Split into 2 parts Back (36x41) falling off Back (32x56) falling off | Back (23x27mm) falling off Back (20x30mm) falling off Back (23x30mm) falling off Scratch-like scar | (25x3Omm) falling off | (25x27mm) falling off (25x32mm) falling off 3 cracks - 60, 66, 30mm | ""Includes blister plate |
|-----------------------|-------------------------------|--|---|--------------------------------------|--|----------------------------------|--|---|-----------------------|---|---|
| Depth | Inden- tation (mm) | 0.8 1.3 | 0.7 | | 0.5 | н | 0.6 | 0.2 | | 1.7 | |
| Diam- | of Hole (mm) | 24x21 25x20 | 12×16 | 32x20 30x20 | 17×12 | | 26x30 21x27 | 17x21 15x24 19x23 13x12 | 20x20 20x20 | 17x19 19x23 | |
| Pene- | e la company | No No Yes Yes | No No Yes No Yes | Yes Yes No | Yes No | No No | No No Yes | Yes Yes Yes Yes | Yes No Yes | Yes Yes No | |
| Corre- | spond- ing Range (m) | 277 27 | 836 963 724 966 918 | 1985 296 392 | 698 772 | 995 996 | 562 395 210 520 | 578 707 772 974 1016 | 798 888 733 | 796 792 820 | |
| Firing | Speed (m/sec) | 514 513 526 526 | 368 339 402 338 348 | 627 580 536 | 411 387 | 333 | 463 535 621 480 | 456 408 387 337 327 | 379 355 399 | 380 381 373 | |
| Type | *TTeus | Common | A. P. | Common | A.P. | Common | | A.P. | Common | Λ.Ρ. | *************************************** |
| et | Thick- ness (mm) | 4.10 4.00 4.00 4.03 | 4.02 4.09 4.13 4.13 | 4.08 4.10 4.05 | 4.07 | 3.82 | 3.8.83 8.82.83 82.82.83 | 3.92 3.92 3.93 3.95 | 3.97 | 3.97 | - |
| Target | No. | 4-1 | | 7-4 | | C-4-1 | | · | 0-4-2 #H | | *Reduced charge |
| Type | | BKg | | | | | | | | | **Reduc |

##Includes blister plate

Projectile head only

 ϕ

9x9

No Head

267

594

A. P.

6.11

ENCLOSURE (A), continued

Center of shell penetrated slight-ly, caused swelling in rear Rear peeled off (4x6)mm Rear peeled off (10x6)mm Caused a slight swelling in rear Caused a slight swelling in rear Caused a round crack-Dia. 30mm Rear peeled off (27x25)mm Rear peeled off (30x20)mm Rear peeled off (26x20)mm Penetration Condition Table V(A)(Continued)
DAIDO STEEL PLATE - RESULTS OF EXPERIMENTS ON PROJECTILE RESISTANCE Inden-tation (mm) 0.0 Depth of n Н Q 7x7 10x10 15x15 front 5x5 6x7 7x6 Diam-eter of Hole (mm) trate Pene-No No Yes No Part. No No Yes Yes No & & ္က မွ 일 일 **윤** 윤 Corre-sponding Range (m) 382 382 338 338 338 338 338 338 35 2282 28 52 288 256 177 284 256 32 Firing Speed (m/sec) **35**5 537 566 555 222 505 5241 607 534 534 575 96 584 599 635 632 599 25 23 23 Common # Common # Common # Common # Type Shell A. P. A.P. A. P. Thickness (mm) 5.8 355 355 **6.12** 6.09 6.09 6.20 6.20 6.20 6.05 6.07 5.98 6.19 6.09 6.15 6.18 6.91 6.16 Target C-6-1 0-6-2 ## 6-2 No. J Type BK3

ry oharge

#Reduced charge #Ordinary charge

**Includes blister plate

Table V(A)(Continued)
DAIDO STEAL PLATE - RESULTS OF EXPERIMENTS ON PROJECTILE RESISTANCE

| Penetration Conditions | Plate broke into triangular form (40x100mm) 5 cracks - 150, 120, 90, 75, 200mm Plate was cut in half | Caused 2 x-shaped cracks in back 120, 110mm | Rear peeled off (25x35mm) 1 crack across 150mm | Rear peeled off(30x32mm) Rear peeled off (40x30mm) 1 crack across 90mm | Broke into many pieces | | | Caused a round crack (75x50mm), 3 cracks on back | n-anapar oo, oo, romi | Back pealed slightly Caused slight crack in back; | Caused Z cracks, 25, 15mm Caused a scratch Caused a scratch | Peeled off (45x45mm) | |
|--------------------------------|---|---|---|--|------------------------|-------------|----------------------|--|-----------------------|---|---|---------------------------------|---|
| Depth of Indentation (mm) | | š0 | 6.5 | | | | | н | 1 | | 0.7 | | |
| Diameter of Hole (mm) | 16x16 | | | 10x13 10x12 | 15x5 | | - - - : | | | 20x1.9 25x20 | | 25x30 | |
| Pene- trate | No Yes | No | N N | Yes Yes | Yes Yes | No | Yes No No | No | No | Yes Yes | No No | No No | No No No Yes Yes |
| Correspond-ing | 188 | 757 | 430 | 346 396 = | 7.6 3.4 | 76 | 480 330 478 | 76 | ଷ | 82 246 | 392 400 337 | 24 | 414 404 318 136 136 185 246 |
| Firing Speed (m/sec) | 630 | 523 | 520 | 557 535 | 7117 | 689 | 497 565 498 | 726 | 729 | 685 595 | 536 533 561 | 720 | 526 531 570 634 631 |
| Type | Соппоп * | A.P. | | | Common # | Common * | A. P. | Coramon # | | A.P. | | Common, | A. P. |
| Thick- ness (mm) | 6.17 | 6.20 | 6.19 | 6.22 | 6.28 6.27 | 6.17 | 6.25 6.25 6.29 | 5.94 | 5.98 | 5.98 | 5.93 | 6.16 6.08 | 6.08 6.15 6.10 6.07 |
| Target I | Ĵ | | | | 5 | le s | | C-6-1 | | | | \$ 5 6 -2 | is . |
| Туре | BK 8 | | | | | | | | | | | | |

*Reduced charge

ENCLOSURE (A), continued

Table VI(A)
DAIDO STEEL PLATE - RESULTS OF EXPERIMENTS ON PROJECTILE RESISTANCE

Date: 8 June 1942 Place: Firing Range of Air Tech. Br. Depot Machine Gun: Vickers, 7.7mm fixed M.G. Type 3, Mod 1

| | | | | | · . | |
|-----------------------|--------------------------|--|---|---|--|---|
| Penetration Condition | | Head only penetrated - caused cracks of 30 and 20mm Penetrated the armor, stopped caused cracks of 65 & 25mm | l crack - 130mm l crack - 30mm Caused a swelling & a crack 40mm Caused a crack 7.8mm | Caused 2 cracks, 15, 15mm Caused only a swelling Caused a crack 30mm Crack 60mm and a peeling off | Plate scattered into several pcs. | 2 cracks - 50 and 45mm 2 cracks - 25 and 20mm 1 crack - 9mm 2 cracks - 80 and 40mm |
| Depth | Inden- tation (mm) | | 6 | 0.6 1.4 | | |
| Size | Hole (mm) | 4x6 | 6x15 30x70 | 6x7 15x52 | 25x10 | 35x35 12x7 |
| Pene- | | Head Part, | No Yes No Yes | Yes Yes Head No No Yes | Head Head No Yea Head No | Yes Yes No Yes |
| Corre | ing Range (mm) | 946 | 670 612 173 48 | 636 970 990 535 382 384 392 | 986 950 954 986 825 942 | 942 960 942 802 |
| Firing | (Des/II) | 342 | 421 443 637 709 | 433 338 334 474 541 540 536 | 335 342 341 335 371 343 | 343 343 378 |
| Angle | Incidence (degrees) | 25 | 5883 | 2222333 | 55.55 | 55 55 55 |
| Type | Tiage | A. F. | A. P. | Q. * | A. * | A. P. |
| et | Thick- ness (mm) | 2.03 | 25.05 8.05 8.05 8.05 | 2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2.2. | 2.23 2.25 2.27 2.23 2.22 2.19 | 2.15 2.17 2.16 2.16 |
| Target | No. | 2-2 | | C-2-2 ## | 2-2 | C-2-2 ## |
| Type | | ВК3 | | : | BK ₈ | |

##Cemented plate

*Reduced charge #Ordinary charge

ENCLOSURE (A), continued

Table VI(A)(Continued)
DAIDO STEEL PLATE - RESULTS OF FROJECTILE RESISTANCE TEST (OBLIQUE FIRING)

| | | | | | | • | | | | | | | | - | |
|----------------------|--------------------------------|---|--|--|---|---|---|---|---|---|--|-----------------------------------|---|---|-------------------|
| State of Penetration | | Scratched Projectile stopped in hole a bulge (13x12mm) peeled off Bulge | (20x27mm) peeled off Little peeled off Little peeled off Little peeled off very near partial penetration | (lOxiomn) peeled off Little peeled off - very near partial penetration | Crack 20mm - little peeled on near surface | little bulge 3 horizontal cracks 50, 50, 30mm | (25x35mm) peeled off 3-30mm cracks 120mm horizontal crack (25x30mm) peeled off- 80, 60, 60, 50mm cracks | On rear surface 120mm horizontal crack | 114tle bulge 90mm credk (35x40mm) peeled off - 100mm horizontal & vertical creck | (30x35) peeled off - 50mm crack 50x25mm torn off | 23x35mm peeled off 30x20mm peeled off Just a scratch | (34x14mm) peeled off - 27mm crack | 2 cracks - 50, 15mm (45x27mm) peeled off (20x18mm) peeled off 30mm crack (27x18mm) - peeled off | Horizontal Louna grack (65x70mm) peeled off | Agreemented plate |
| Depth of | tation (mm) | 1,4 1,6 1,2 | 9 | | io. | W N | 1.00 | н | 1.7 | | 1.5 | | 0.0 | 1,2 | |
| Size | Hole (mm) | 7×12 | 12x12 15x9 7x13 7x6 | 7x13 7x6 | 7×12 | - 61x7 | 15x23 | | 23×30 | 19x20 | 16×19 1,5×22 | *77x02 | 23x23 15x14 17x20 | 40x50 | - |
| Type of Penetra- | tion | No No Part No Yes No | Yes Yes Yes Head | Yes | Yes | No No Yes | Yes No No Yes | oN. | No No Yes | Yes | Yes Yes No | Yea | No No Yes Yes | No | |
| Range (meters) | # | 990 1006 684 430 337 370 | 370 446 447 576 | 556 576 | 158 | 2000 2000 2000 2000 2000 2000 2000 200 | 717 698 656 601 | 322 | 280 166 60 | 011 | 316 446 995 995 | 7z9 | 1000 166 302 346 | 100 | |
| Impact Veloc- | ity (m/sec) | 45 52 53 34 52 53 54 54 54 54 54 54 54 54 54 54 54 54 54 | 546 530 512 457 | 465 | 111 9 | 583 597 624 | 83 117 82 147 147 | 568 | 588 640 700 | 199 | 57. 33. 33. 33. 33. | 1,38 | 332 | 22,8 | |
| Firing | <u>.</u> | 25 25 45 45 45 45 | 8.8.8.8 | 25 | 57 | 75 72 72 72 72 | 3333 | 45 | 45 45 45 | 45 | 8888 | <u>۾</u> ۾ | 30 30 30 30 30 30 30 30 30 30 30 30 30 3 | 55.5 | |
| Type Shell* | | A. * | G. * | | | | A * | | | A. P. | * | | | | - |
| Target | Thickness Where Hit (mm) | ~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~~ | 7.4.4.4.7.7.7.7.7.7.7.7.7.7.7.7.7.7.7.7 | 4.18 | 4.17 | 4.14 4.07 4.18 | 4.01 4.04 4.03 | 60*7 | 40.4 40.4 40.4 | 4.01 | 3.85 | 3.93 | 888888 | #88 101 | 4.07 |
| Į. | No. | 2-4 | 2 7 7 8 | - | | | 4-2 | | | | C-4-2 | | | | |
| Type | | BK ₃ | | | | | BK 8 | | 21 | | · · · · · · · · · · · · · · · · · · · | | | | |

"Reduced charge #Ordinary charge

ENCLOSURE (A), continued

| | | | | | | a | | | | |
|----------------------|--------------------------------|--|--------------------------|----------------------|---|---|--|----------------------|---|------------------|
| State of Penetration | | Back surface cracked 20x26mm peeled off | * | * | Rear surface peeled off Scratch Scratch | 25x25mm peeled off 25x25mm peeled off 25x25mm peeled off 30x35mm peeled off 25x25mm peeled off 3 cracks - 360, 320, 180mm small pieces flew off Flew off in several pieces | 30x35mm peeled off 30x30mm peeled off | | 30x45mm peeled off Scratches Scratches Scratches | ##Cemented plate |
| Depth | Inden- tation (mm) | m m | 2 | , K | | | | | | |
| Size | Hole (mn) | 7x7 13x16 | 10x16 | 8 x1 0 | 9x1.5 | 12xd5 12xd5 12xd5 12xd5 12xd1 | 15x7 15x20 | 27x27 28x30 | 10x20 | |
| Type of Penetra- | tion | Head No Yes No | Yes No No | No Yes No | Yes No No | Yes Yes Yes Yes No No | Yes Yes No | Yes No Yes | Yes No No | |
| Range (meters) | 11 T | 158 171 146 181 | ⊉ &%% | 158 85 100 | 24 104 294 | 185 160 160 184 184 184 184 184 184 185 185 185 185 185 185 185 185 185 185 | 27.2 | 154 158 44 | 54 64 62 48 | |
| Impact | ity (m/sec) | 647 636 649 649 | 712 678 706 706 | 644 683 673 | 774 704 673 678 | 631 643 652 632 605 605 | 722 714 705 | 646 644 712 | 705 698 699 709 | |
| Firing | (<u>o</u>) | rv rv og | 8 25 8 23 | ୡୡୡ | 82838 | 8 84446 | 25 30 35 | 258 | 25 30 35 40 | |
| Type | | ₽. * | A. P. | A.P. | A.P. | * * | A.P. | A. P. | A.P. | |
| Target | Thickness Where Hit (mm) | 5.97 5.97 5.97 | ~~~~ \$\$\$\$ | 6.20 6.23 6.17 | 6.23 6.10 6.28 6.26 | 44.44.44.44.44.44.44.44.44.44.44.44.44. | 6.21 6.21 6.26 | 6.09 6.09 6.09 | 6.08 6.15 6.07 6.09 | Ţ. |
| Ta | No. | 1–9 | ## . | c-6-2 ## | | 2-9 | | C-6-2 ## | 4. | *Reduced charge |
| Type | | EK3 | | | | BK 8 | | ` c | | #Reduce |

*Reduced charge //

Table VII(A) BALANCED FIRING ANGLE

| Туре | Thickness (mm) | Treatment | Angle | Note |
|-----------------|----------------|------------|-----------------|-----------|
| ę. | | | 60° | A |
| | 2 | Carburized | 58° | Estimated |
| , | , | | 54° | Estimated |
| BK ₃ | 4 | Carburized | 50° | Estimated |
| , | , | | ц° | |
| | 6 | Carburized | 18° | · |
| | 2 | | | |
| | 2 | Carburized | | |
| DV | , | | 47° | |
| BKg | 4 | Carburized | 55 ⁰ | Estimated |
| | 6 | | 29 ⁰ | |
| | O | Carburized | 20° | 3 |

Table VIII(A)
RESULTS OF PROJECTILE RESISTANCE TEST
A.P. SHELL (ORDINARY CHARGE)

| Types of Targets | Thick- ness of Plate (mm) | Impact Velocity (m/sec) | Range (m) | Size of Hole (mm) |
|--|--|---|--|--|
| BK3 2-1 | 2.11 | 771 | 90 | 15x18 18x13 18x19 15x15 17x18 15x19 15x15 |
| BK3 2-2 | 2.02 | 760 | 126 | |
| BK3 C-2-1 | 2.49 | 764 | 112 | |
| BK3 C-2-2 | 2.48 | 770 | 92 | |
| BK8 2-1 | 2.25 | 767 | 105 | |
| BK8 2-2 | 2.24 | 763 | 116 | |
| BK8 C-2-1 | 2.10 = | 754 | 158 | |
| BK8 C-2-2 | 2.20 | 764 | 112 | |
| BK3 4-1 BK3 4-2 BK3 C-4-1 BK3 C-4-2 BK8 4-1 BK8 4-2 BK8 C-4-1 BK8 C-4-2 | 4.08 3.96 4.25 4.05 4.09 4.10 3.82 4.00 | 760 771 765 758 764 767 775 | 126 90 110 130 112 104 78 140 | 18x13 15x18 14x21 18x18 24x20 22x18 30x20 14x16 |
| BK ₃ 6-1 | 5.93 | 767 | 104 | 22x38 |
| BK ₃ 6-2 | 6.09 | 765 | 110 | 26x26 |
| BK ₃ C-6-1 | 6.07 | 779 | 66 | 24x27 |
| BK ₃ C-6-2 | 5.25 | 762 | 118 | 20x22 |
| BK ₈ C-6-1 | 5.95 | 766 | 106 | 25x23 |
| BK ₈ C-6-2 | 6.08 | 771 | 90 | 25x25 |

Note: All plates were penetrated.

Date: 8 June 1942 Weapon: Experimental (Air) 13mm fixed machine gun Place: Firing Range of Branch Air Tech. Depot

Table X(A)
DISTRIBUTION OF HARDNESS

| Type | Surface | Plate | Hardness | (Rockwell) |
|------|----------------|--|---|--|
| | Treat- ment | Classi- fication | Cemented Surface | Rear Surface |
| | | 2-1 2-2 4-1 4-2 6-1 6-2 | 55 , 54 , 53.5, 53.5 53 , 54 , 53 , 53 53 , 54 , 53.5, 52 53 , 53 , 52.5, 52 51 , 54 , 54 , 51 50.5, 51.5, 52 , 53 | 54 , 53.5, 53.5, 52 51.5, 50 , 53 , 51 51.5, 52 , 53 , 52.5 51 , 53.5, 52.5, 52 50.5, 50.5, 52 , 52 51 , 51 , 53.5, 53 |
| BK3 | Cemented | C-2-1 C-2-2 C-4-1 C-4-2 C-6-1 C-6-2 | 51.5, 52 , 53 , 53 50.5, 52 , 53.5, 53 52.5, 53 , 52.5, 53 54 , 52.5, 51 , 53 57 , 58 , 58 , 57 56 , 57 , 57 , 57 | 51 , 50 , 50.5, 50.5 53 , 52.5, 52 , 53 52.5, 50.5, 50 , 51.5 50 , 50.5, 51 , 51 49.5, 49 , 50 , 50.5 51 , 52 , 51 , 52 |
| | | 2-1 2-2 4-1 4-2 6-1 6-2 | 55.5, 55 , 55 , 55 55 , 56 , 56.5, 57 55 , 53 , 53 , 54 56 , 53 , 55 , 55 56 , 57 , 57.5, 58 56 , 56 , 56 , 56 | 55 , 54 , 54 , 54 54 , 55.5, 56 , 55 55 , 53.5, 55 , 53 53 , 53.5, 54 , 53.5 56 , 56 , 53.5, 55 55 , 55 , 53.5, 53 |
| BK8 | Cemented | C-2-1 C-2-2 C-4-1 C-4-2 C-6-1 C-6-2 | 56 , 57 , 56 , 55.5 56 , 54 , 55 , 56 55.5, 54 , 53 , 54 58.5, 65 , 58 , 59 56 , 58 , 57 , 57 58 , 57 , 57 | 54 , 53 , 54 , 53 55.5, 55 , 54 , 53 54 , 53.5, 53.5, 54 53 , 51 , 51 , 53.5 54 , 52 , 53.5, 52 56 , 56 , 55.5, 55 |

Points Where Hardness is Measured

Table XI(A)
CELENTED THICKNESS

| Туре | Cemented Thickness in Percent | | | | | | | | |
|-------------------|-------------------------------|--------------|-----------|--|--|--|--|--|--|
| | 2mm Plate | 4mm Plate | 6mm Plate | | | | | | |
| ВК3 | | | 16 | | | | | | |
| вк _в " | 15 | 13 (Approx.) | 15 | | | | | | |

Table XII(A)
NON-METALLIC INCLUSIONS
(Gakushimho)

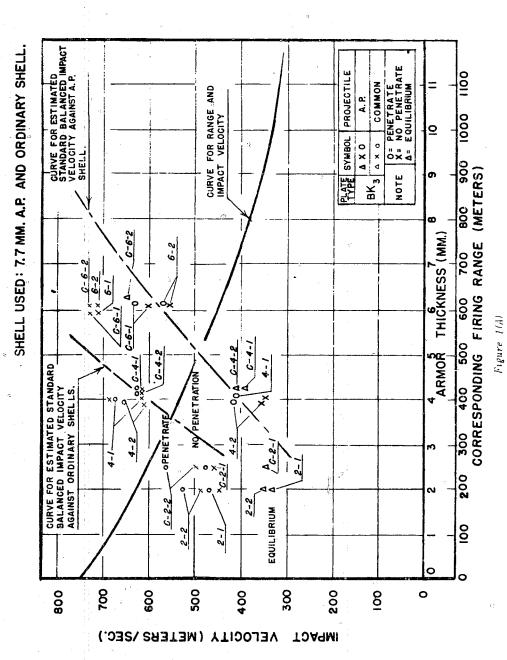
| Туре | Impuri- ties Number | Distribu- tion Ratio (a) | Number in Field of Vision | бхв | Degree of tab | Non-me Obje | tallic cts |
|-----------------|------------------------------|--------------------------------|---------------------------------|-------------------------------|------------------|----------------|------------------------|
| | Namosi (a) | | (b) | | Purity | (c)* | € <u>abc</u> ** €ab |
| g | 0 0 1 1 2 3 4 4 8 5 16 | | 34 14 2 0 0 | 0 14 4 0 0 | <u>18</u> 50 | 6 3 | 4.5 |
| вк3 | Tot | al | 50 | 18 | =0.36 | | |
| | 0 0 1 B 2 2 3 4 8 5 16 | | 10 20 20 0 0 | 0 20 40 0 0 0 | <u>60</u> 50 | 4 4 | 4 |
| | Tot | al | 50 | 60 | =1.2 | | |
| | 0 1 2 3 4 5 | 0 1 2 4 8 16 | 25 0 12 11 2 0 | 0 0 24 44 16 | <u>84</u> 50 | 4 3 3 | 3 |
| BK ⁸ | Total | | 50 | 84: | =1.68 | | |
| 11 | 0 1 B 2 3 4 | 0 1 2 4 8 1ú | 18 0 6 20 6 | 0 0 12 80 48 C | <u>140</u> 50 | 3 3 3 | 3 |
| | Tot | al | 50 | 140 | =2.8 | | i |

^{*}Average Thickness

Table XIII(A)
MEASUREMENT OF CRYSTALLIZATION
(Gakushinho)

| Туре | Degree of Crystal- ization (a) | Number in Field of Vision (b) | axb | €ab* €b | Estimate |
|------|---|--|------------------------|------------|----------|
| BK3 | G 5 G 6 G 6.5 G 7 | 2 3 3 2 | 10 18 19.5 14 | 61.5 10 | 6.15 |
| | Total | 10 | 61.5 | =6.15 | G 6.15 |
| BKg | G 6.5 G 7 G 7.5 | 5 3 2 | 32.5 21 15 | 68.5 10 | 6.85 |
| | Total | | , 68.5 | =6.85 | G 6.85 |

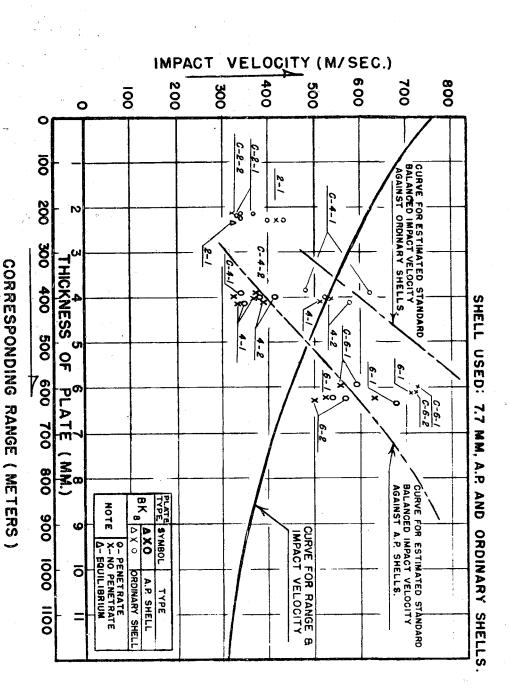
^{*}Average Degree of Crystallization



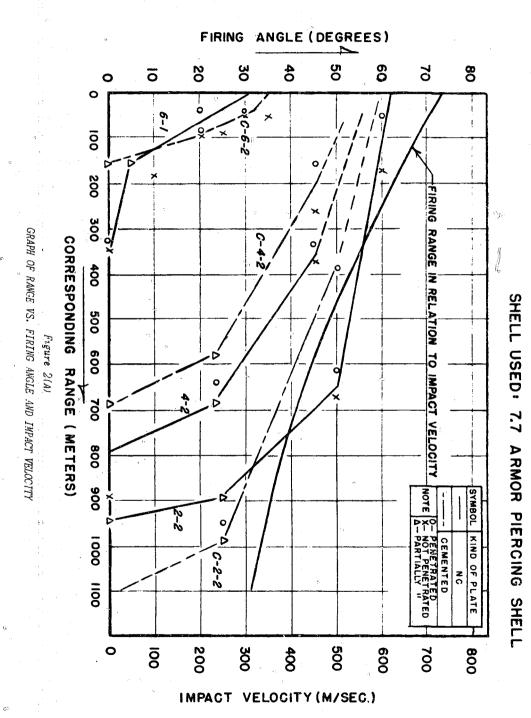
BALAWOED THPACT VELOCITY AND RAWGE CHRUE

BALANCED IMPACT VELOCITY AND RANGE CURVE

Figure 1(A) Cont.



45



46

FIRING ANGLE

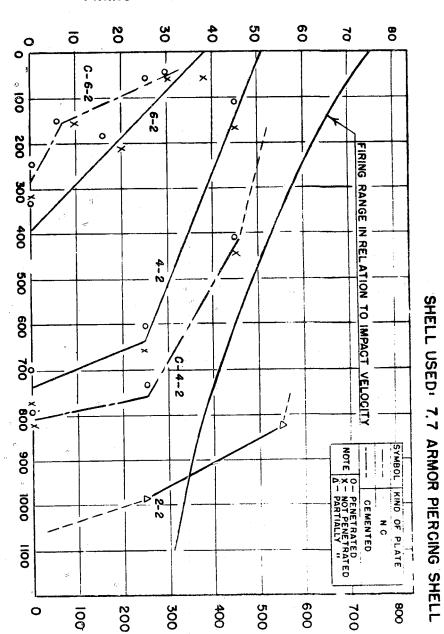


Figure 2(A) Cont.
GRAPH OF RANGE VS. FIRING ANGLE AND IMPACT VELOCITY

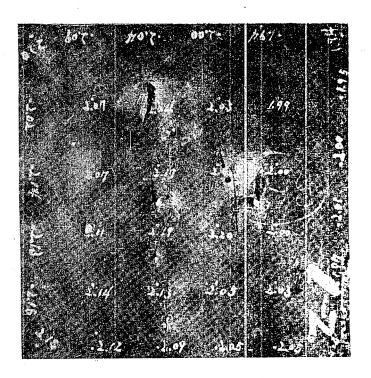
CORRESPONDING RANGE (METERS)

47

IMPACT VELOCITY

M/SEC.

ENCLOSURE (A), continued



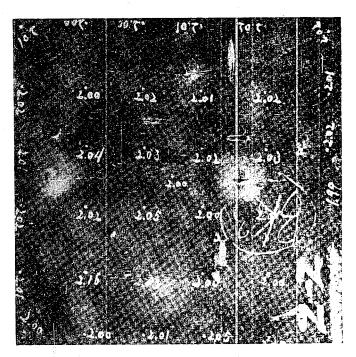
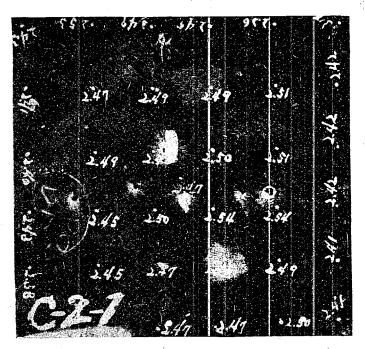


Figure 3(A)
BKo --- 2mm PLATE

ENCLOSURE (A), continued



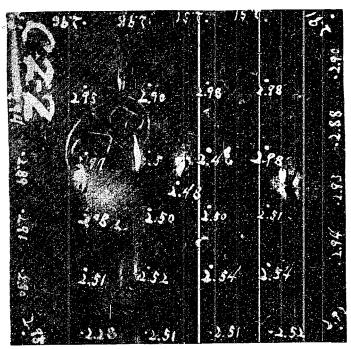
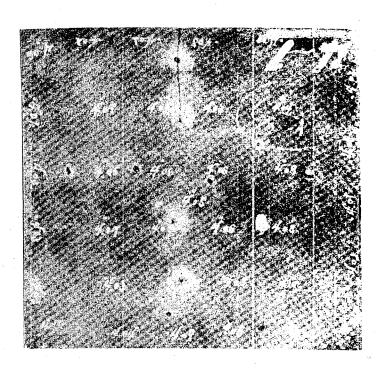


Figure 3(A) Cont.

BKs --- Zum (CEMENTED) FLATE



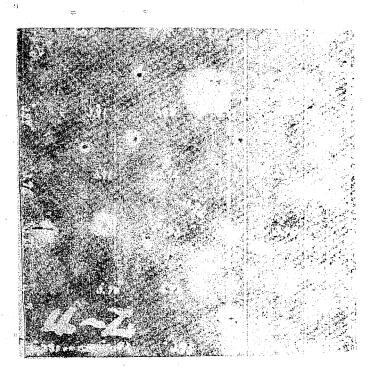
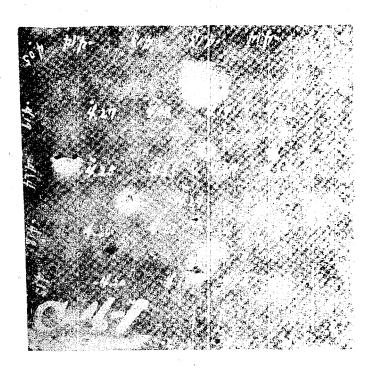


Figure 3(A) Cont.

BK₃ ——— 4mm PLATE

ENCLOSURE (A), continued



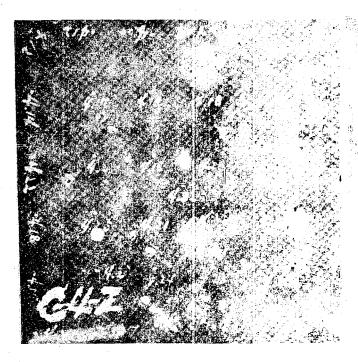
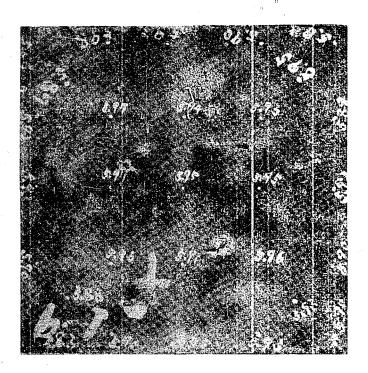


Figure 3(A) Cont.

BKs —— 4mm (CEMENTED) PLATE

ENCLOSURE (A), continued



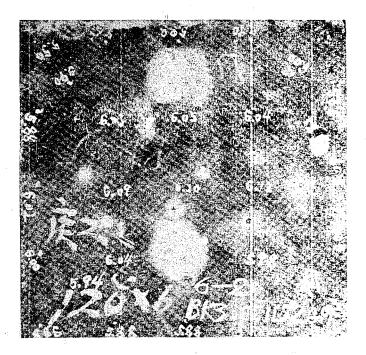
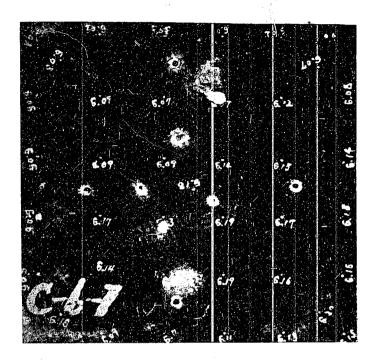


Figure 3(A) Cont.

BK₃ --- Omm PLATE

ENCLOSURE (A), continued



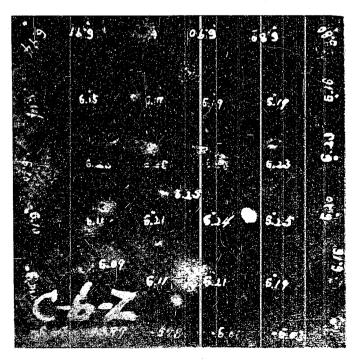
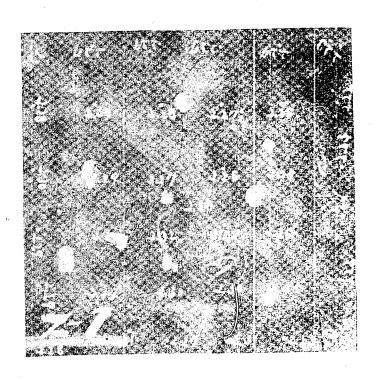


Figure 3(A) Cont.

BK3 --- Gram (CEMENTED) PLATE

ENCLOSURE (A), continued



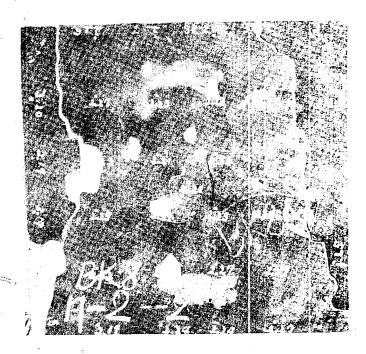
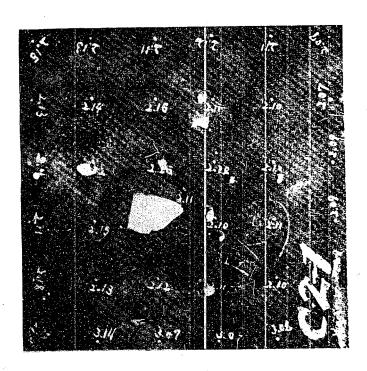


Figure 4(A)
Bha --- 2mm PhATE

ENCLOSURE (A), continued



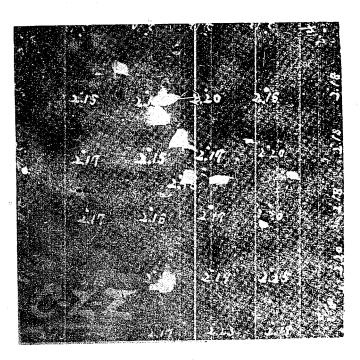
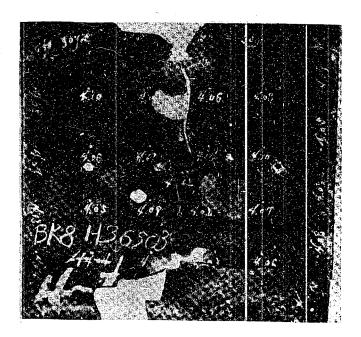


Figure 4(A) Cont.

BKs --- 2mm (CEMENTED) PLATE

ENCLOSURE (A), continued



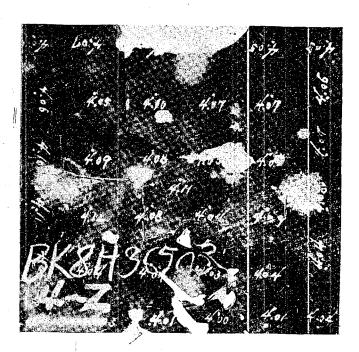
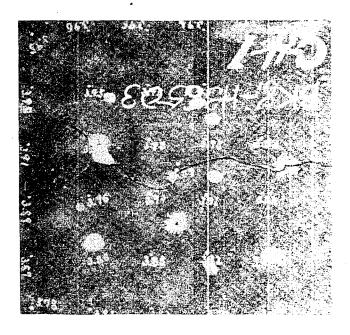


Figure 4(A) Cont.

BK8 --- 4mm PLATE

ENCLOSURE (A), continued



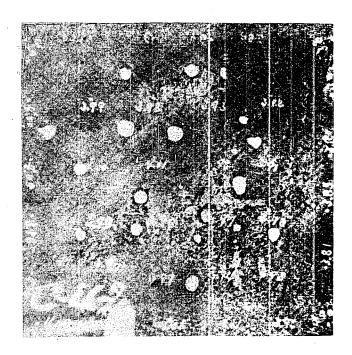


Figure 4(A) Cont.

BKs --- Amm (CEMENTED) PLATE

ENCLOSURE (A), continued

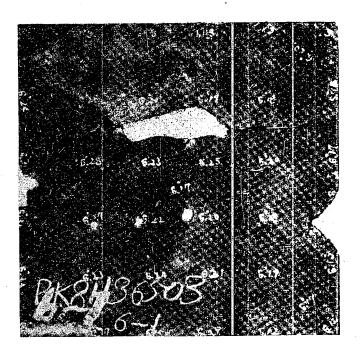
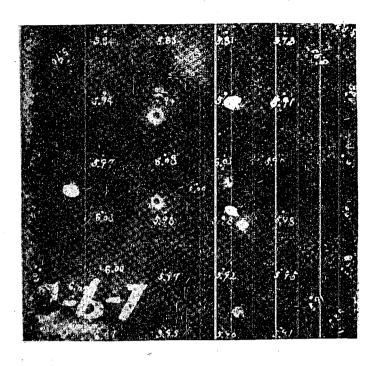




Figure 4(A) Cont.

EKa --- Orm PLATE

ENCLOSURE (A), continued



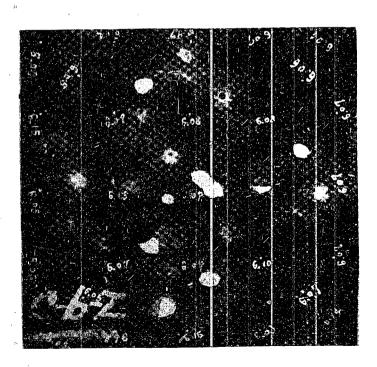


Figure 4(A) Cont.

BK8 --- 6mm (CEMENTED) PLATE

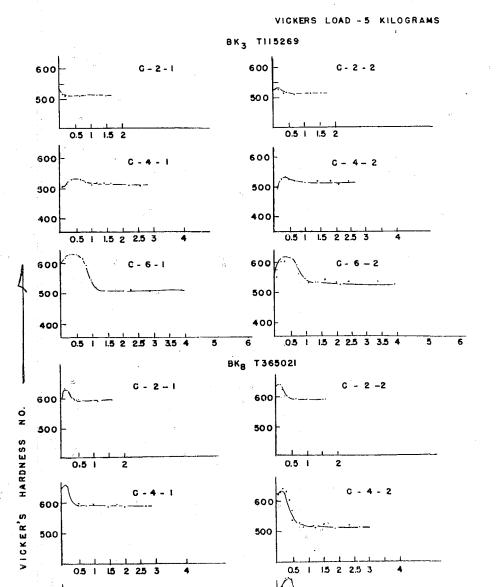


Figure 5(A)

CROSS-SECTION HARDNESS CURVES

600

500

0.5 1 1.5 2 2.5 3 3.5 4

PLATE DEPTH MM

C - 6 - 1

0.5 1 1.5 2 2.5 3 3.5 4

600

500



Figure 6(A)
NON-METALLIC FOREIGN INCLUSIONS

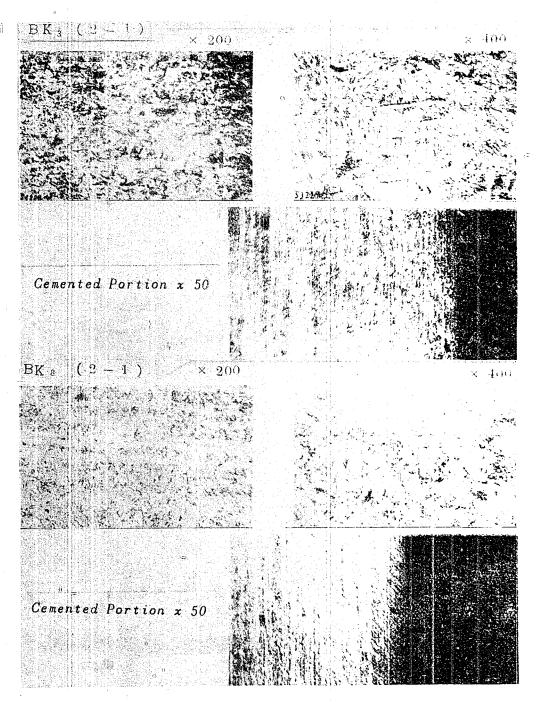
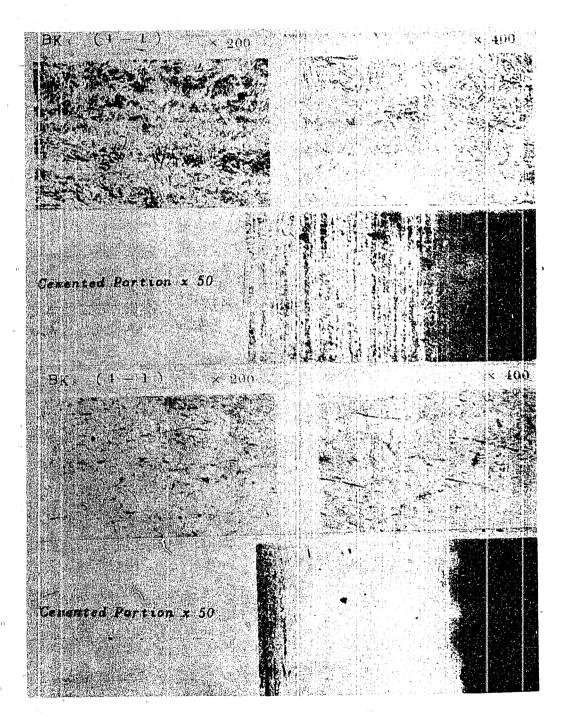


Figure 7(A) F MICROSCOPIC FORMATIONS



• Figure 7(A) Cont.

MICROSCOPIC FORMATIONS

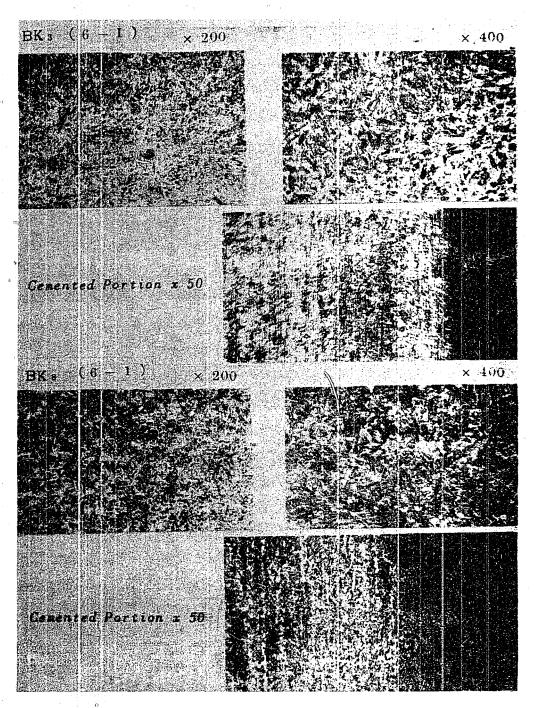


Figure 7(A) Cont.

MICROSCOPIC FORMATIONS

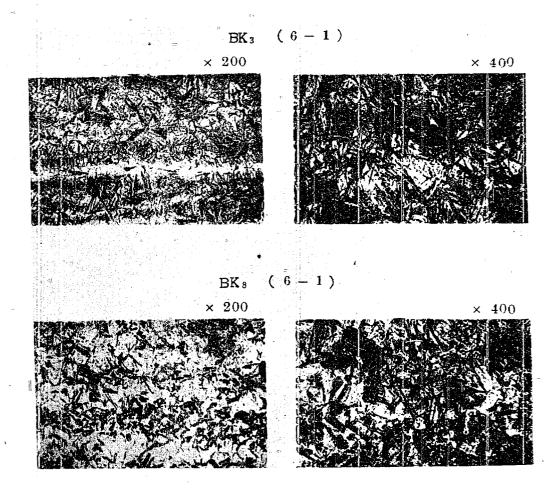


Figure 8(A)
MICROSCOPIC FORMATION WHEN ANNEALED

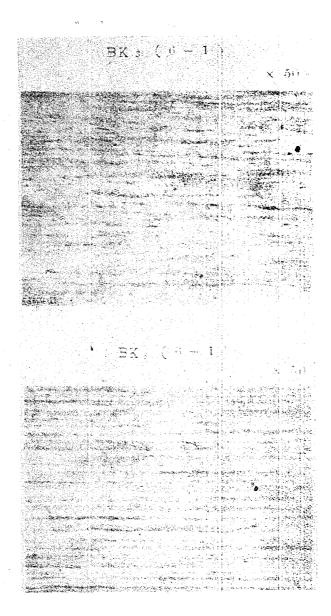


Figure 9(A)

BINDING AND BREAKING

ENCLOSURE (A), continued

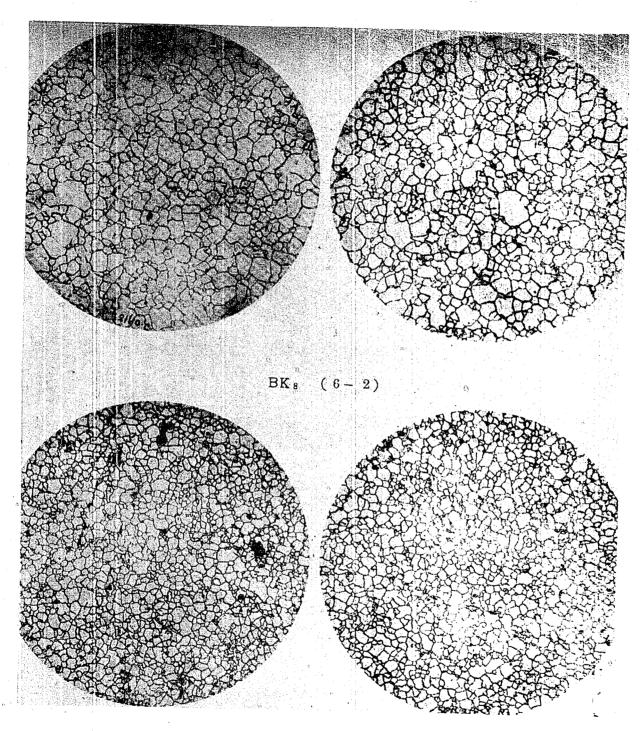
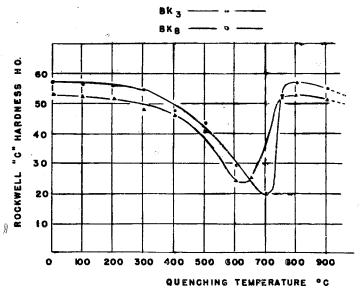


Figure 16(A)

DEGREE OF CRYSTALLIZATION

QUENCH IN OIL EVERY HOUR



HARDENING - TEMPERING CURVE

Figure 11(A)
HARDENING-TEMPERING CURVES

ENCLOSURE (B)

DISCUSSION OF THE TORSION NON-BALLISTIC TEST

A. SUMMARY

- 1. At the Metals Research Institute in SENDAI, Japan, Professors MURA-KAMI and WATANABE, under the direction of Dr. HONDA, president of the university at SENDAI, developed a technique in which the ballistic quality of light steel armor plate against a 6.5mm rifle bullet could be predicted quantitatively by the use of a static torsion test. While the application of the test is somewhat limited and, because of the Allied occupation, additional results on other calibers and with different test conditions could not be obtained, it is considered that the technique merits consideration.
- 2. It was found that the torsional energy per unit area (integral of torsional stress vs. twist angle) for 50° of twist was inversely proportional to the limiting thickness of armor made from the same steel which would just resist the penetration of the bullet at normal obliquity at a constant standard velocity. When the torsion specimen broke below 50° twist, it usually did so with a helical granular fracture on a 45° plane. In such a case, the bullet would shatter the plate (or at least cause sections of plate to fly off). When the specimen broke after being twisted through an angle greater than 50°, it did so with a smooth fibrous fracture in a plane normal to the axis. (At angles near 50°, there was an occasional inconsistency or a fracture whose nature was a hybrid.)

B. LIMIT DETERMINATION

- 1. The limiting thickness of a given steel with a given heat treatment was determined by firing about five such plates of various thicknesses less than the limit. The residual velocity of the bullet was measured, and only two rounds usually were necessary on each plate. The residual velocity was then plotted against thickness and the curve extrapolated to zero residual velocity. Naturally, after the first one or two tests, it was possible to select thicknesses quite close to the limit, thus reducing to a minimum the error involved in extrapolating. The thickness at zero velocity was known as the critical gage.
- 2. The bullet used had a 6.5mm diameter and weighed 9 grams. It had a lead core in a cupro-nickel jacket. The mean velocity was about 760 meters/second, a rather high velocity, but this permitted more shock, closer control, and less proportional error. The muzzle was only about 10 feet from the plate.
- 3. The plates were prepared in various manners as discussed below. They were cut to about 9" x 9" and secured to a massive frame. A half dozen diaphrams, plates with small holes in the middle, were lined up behind the frame. These diaphrams permitted free passage of the bullet but screened out most of the plate fragments if any were thrown back. The bullet finally entered a ballistic pendulum about two feet behind the plate.
- 4. The pendulum was an ordinary type, suspended by rigid frames fore and aft. It swung free in the line of fire so that the recording apparatus on the side described an arc on the paper. The bullet was caught in a sand box. It penetrated a thin wood bulkhead and a layer of thin rubber which had self-sealing properties to prevent the sand from leaking out. The weight of the box was carefully controlled.
- 5. The pendulum had been calibrated so that the length of the described

arc could be converted directly to velocity. The lever arm of the pendulum was about three feet. The weight of the box is available in documents on the subject (see list of Japanese documents pertaining to light armor, Enclosure E) but was not discussed during interrogations.

6. The residual velocity was plotted against the plate thickness (which was obtained at will by varying the spacing of a small rolling mill), and the curve extrapolated as shown in Figure 1(B).

C. TORSION TEST

- l. One of the objections to this experiment was the fact that torsion specimens could not be taken directly from the plates. The plates were always too thin. However, the specimens were obtained from the same heat of steel as the plates and were heat-treated along with the plates. They were machined after normalizing and were merely ground to accurate dimensions after the quench and temper. The shafts of the specimens were 10mm in diameter and about 10mm long with a generous fillet at each end. The specimens had cylindrical heads which were flattened on one side to fit the chucks on the machine. After the test, the Rockwell hardness was checked against that of the plate to prove similitude.
- 2. The machine was motor-driven and required roughly seven minutes to turn through 100°. The end opposite the power-end was suspended in frictionless bearings and bore a heavy pendulum which was forced upwards by the torque on the specimen. The angle of its rise indicated the moment of torque. The record was obtained by a drum which turned according to the twist angle, and a stylus which moved laterally according to the moment of torque (or torsional shear stress).
- 3. Three typical curves are shown in Figure 2(B). The solid line (b) indicates a satisfactory good steel. The sharp peaked line (a) shows a break at a low twist angle indicating brittleness which would result in shattering on ballistic test. The low long curve (c), indicates a plate too soft, having a low penetration resistance. The integreated areas below 50° twist for (c) are shaded in figure 2(B). The area under curve (b) from 0° to 50° is obviously larger than that for curve (c). Even though the area under curve (a) from 0° to the breaking angle may be satisfactory, the plate would exhibit brittle properties on a ballistic

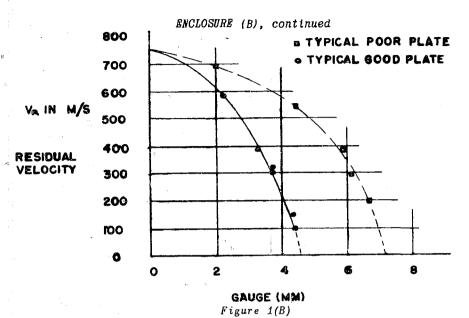
test. The area $\int_0^{\infty} S_T d\phi$ is the torsional energy per mm² required to twist the specimen through 50°. A minimum stress of 150 kg/mm² was considered desirable for good penetration resistance.

D. THE CORRELATION

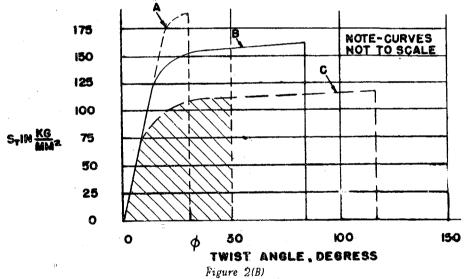
1. The results of the tests discussed above were plotted for many mediumalloy steels with various heat treatments. The chart was similar to Figure 3(B). The data plotted with a band, as shown, and concentrated on one straight line. The most satisfactory plates had a critical gage limit of 4.5mm for the test conditions used. The above plot holds only for medium alloy, tempered, martensitic steel. It would not hold for austenitic steel, for example.

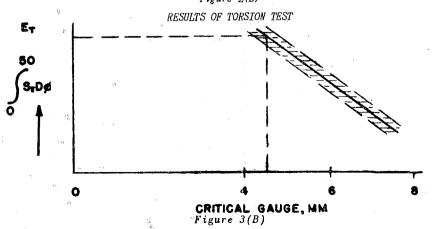
E. APPLICATION

1. The Japanese had applied the correlation discussed above to a series of tests in which they determined the optimum compositions and tempering temperatures for substitute alloys containing no nickel. A large number of specimens were tested in the torsion machine and a few finally emerged as satisfactory, all without firing a shot. A given analysis was tested



VELOCITY - GAUGE CURVES FOR 6.5mm BULLET TESTS





CORRELATION OF TORSIONAL ENERGY AND CRITICAL GAUGE

at various tempering temperatures. The plots of maximum torsion angle versus tempering temperature were interesting in that they showed two distinct dips in many cases, one at a low temperature in the order of 175°C, and one about 550°C. This demonstrated the existence of two zones of temper embritlement. The low temperature zone is not manifested on subsequent low temperature draws, but only on the initial tempering from the martensitic structure.

- 2. The purpose of this discussion is not to present the data obtained by the Japanese, since the application of these data would be limited. Rather, it is to present a technique which might be useful.
- 3. Further work had been started using 7.7mm steel bullets and work was planned with a 20mm AP projectile. The 20mm test required a special larger firing range, and it had not progressed to the point of obtaining any data when the war ended.
- 4. The following compositions and heat-treatments of steels gave good results on the torsion test.

| C+ 1 5 | | Coi | npositi | on (%) | | | | I | leat Tr | eatment | |
|----------|------------------------------|--------------|------------------------------|------------------------------|----------------------|------------------------------|----------------------------------|--------------------------|-------------------|--|-------------------------|
| Steels | C | Cr | Mn | Mo | v | Si | Qu | enche | đ - | Tempered | |
| | | | 4 | | | | °c | Time | Cool- ing | Range (°C) | Time |
| Cr-Mo° | 0.45 0.41 | 1.82 1.74 | <0.4 | 0.59 0.82 | | <0.2 <0.2 | 880°C 850°C | 301 301 | oil oil | 100°-150° 100°-150° | lhr. lhr. |
| Cr-Mn-Mo | 0.39 | 1.50 | 1.31 | 0.76 | | < 0.2 | 880°C | 301 | air- cool | 250° - 300° | i |
| lin-lio | 0.36 0.46 0.37 0.37 | | 1.70 1.87 1.69 1.70 | 0.52 0.63 0.31 0.61 | | <0.2 <0.2 <0.2 <0.2 | 860°C 860°C 860°C 860°C | 301 301 301 301 | oil oil oil | 100°-150° 100°-150° 100°-150° 100°-150° | 3hrs. |
| Mn-V | 0.41 0.40 0.31 | J | 1.61 2.27 1.68 | | 0.22 0.22 0.23 | <0.2 | 0,088 0,088 0,088 | 30' 30' 30' | oil oil oil | 150° 150° 150° | 3hrs. 3hrs. 3hrs. |
| Mn-Si | 0.34 0.42 0.48 | | 1.51 1.51 1.60 | | | 1.03 1.09 0.23 | 880°C 880°C 880°C | 301 301 301 | oil oil oil | 200 ⁰ -250 ⁰ 150 ⁰ 150 ⁰ | 3hrs. 3hrs. 3hrs. |

ENCLOSURE (C)

REDUCTION OF SLAB TO FINAL THICKNESS AS PERFORMED BY THE DAIDO STEEL MANUFACTURING CO., NAGOYA

A brief description of the process of reducing an ingot to the final thickness of armor at the Daido Steel Manufacturing Co., Hoshizaki, NOGOYA is given in the following outline:

I. Reducing Ingot to Slab

- A. Size of ingot, see Figure (C)1.
- B. Ingot forged by steam hammer to slab of approximate dimensions, 60-70mm x 250mm x 900mm.
- C. The rolling of the slab to a 30mm thickness for carburizing is given in Figure 2(C). The standard draft per pass averaging about 10% of the thickness.

II. Carburizing

- A. Sizing: Finished slab 30mm x 450mm x 1000mm is cut to 30mm x 390mm x 930mm for carburizing.
- B. The slab is carburized for 100 hours at 900-950°C in a box as shown in Figure 3(C); the depth of carburization averaged about 20%.

III. Rolling Carburized Plate to Final Thickness

A. Rolling procedure: The carburized plate 30mm x 390mm x 930mm is rolled first to the desired width in one direction, then rotated 90° and rolled to the desired gauge. Sizes (mm) of production armor plate were as follows:

| THICKNESS | <u>WIDTH</u> | <u>LENGTH</u> (as rolled) |
|-----------|--------------|------------------------------|
| 16 | 800 | 930 |
| 16 12 | 450 800 | 1500 1170 |
| 8 | 800 | 1750 |
| 6 | 800 | 2470 |

B. The number of passes required to obtain specific thickness (mm) was as follows:

| | No. of Passes | | | | | | | | | | | | | |
|-----------|---------------|----|-----|----|----|----|----|----|----|----|----|----|----|----|
| | 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 13 | 14 |
| Thickness | 30 | 27 | 24 | 21 | 19 | 18 | 16 | 14 | 12 | 10 | 9 | 8 | 7 | 6 |
| Draft | 3 | 3 | . 3 | 3 | 2 | 1 | 2 | 2 | 2 | 2 | 1 | 1_ | 1 | 1 |

C. Calculated from the surface area, the following tables give the ratic of cross-rolling in reducing the carburized slab to a desired thickness and width.

ENCLOSURE (C), continued

| Cross-Rolling Increase in Area and Ratio | | | | | | |
|--|-------------------------------------|-----------------------------|-----------------------|--|-----------------------------|-----------------------|
| Original Plate Size (mm) | Size After First Rolling (mm) | Increase in Area (mm) | Roll- ing Ratio | Size After Second Roll- ing (mm) | Increase in Area (mm) | Roll- ing Ratio |
| 30x390x930 | 26x450x930 | 60 x 9 3 0 | 1 | 16x450x1500 | 450x570 | 4 |
| 30x390x930 | 16x800x930 | 410x930 | 1 | | | |
| 30x390x930 | 16x800x930 | 410x930 | 2 | 12x800x1170 | 800x240 | 1 |
| 30x390x930 | 16x800x930 | 410x930 | 1 | 8x600x1750 | 800x820 | 2 |
| 30x390x930 | 16x800x930 | 410x930 | 1 | 6 x 800x2470 | 300x1540 | - 3 |

IV. Discussion

From the above data it may be noted that cross-rolling was accomplished on all plates manufactured by Daido and that the amount of cross-rolling was dependent upon the desired thickness.

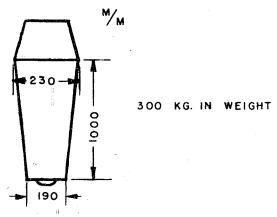
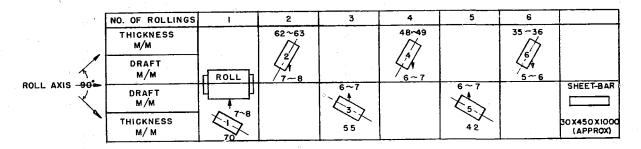


Figure 1(C)
INGOT



REMARKS

- 1. THESE ROLLINGS ARE CARRIED ON BY ONE HEAT.
 2. AT EACH ROLLING THE PLATE IS PASSED IN
 THE DIRECTION TURNED AT 45° TO THE ROLL AXIS: EACH ROLLING IS PERFORMED TURNING 90° SUCCESSIVELY.
- 3 AMOUNT OF CROSS ROLLING IS NATURALLY 50%.

Figure 2(C) METHODS OF ROLLING

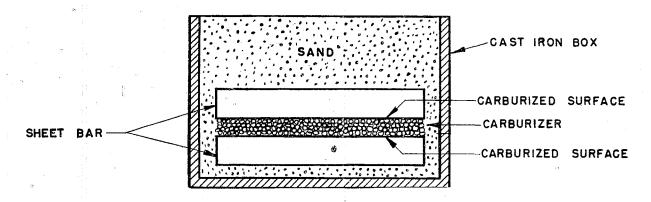


Figure 3(C) CARBURIZING BOX

ENCLOSURE (D)

LIST OF JAPANESE LIGHT ARMOR SHIPPED TO NAVAL PROVING GROUND, DAHLGREN, VA.

| NavTechJap No. | Type | Thickness | No. of Plates |
|----------------|----------------------------|-------------------------------|---------------|
| JE50-3142 | Homogeneous | $3/8$ in to $1\frac{1}{2}$ in | 6 |
| JE50-3143 | Ducol (High Mn) | 3/16 in to 1 in | 8 |
| JE50-3144 | Carburized- Homogeneous | $3/8$ in to $\frac{1}{2}$ in | 9 |

ENCLOSURE (E)

LIST OF JAPANESE DOCUMENTS PERTAINING TO LIGHT ARMOR FORWARDED TO WASHINGTON DOCUMENT CENTER

| NavTechJap No. | ATIS No. | <u>Title</u> |
|----------------|----------|--|
| ND50-3165 | 4120. | Research on Rifle Shields, 1937 |
| ND50-3166 | 4121 | Firing Tests on Non-Nickel Shields, 1943 |
| ND50-3163 | 4118 | Study of Trajectory of a 5cm AP Projectile when Firing against Armor, 1943 |
| ND50-3146 | 4115 | Measurement of Penetration Resistance of Shield Plate versus Rifle Bullet, 1931 |
| ND50-3162 | 4117 | Experiment Mn-Va Steel Torsion Test of 1944 |
| ND50-3164 | 4119 | Measurement of the Penetration Resistance of Armor with Ballistic Pendulum, 1932-1943 |
| ND50-3167 | 4122 | Experiments of Special Steel Plates, 1940 |
| 4 ND50-3147 | 4116 | Bullet Proof Glass, 1939-1943 |
| ND50-B157 | 3833 | 25mm AP Projectiles versus Thin Armor. Manufacture & Ballistic Test of Composite Armor |
| ND50-3189 | 4380 | Experiments on Pressure Resisting Qualities of Glass in Lookout Windows |